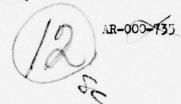


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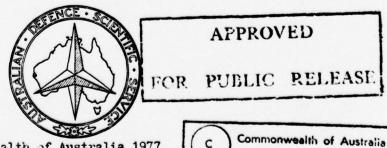
FIN STABILISED BOMB MODEL

M.Pollock, R.G. Broadbent & L.J. Roberts

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TRANSONIC WIND TUNNEL TESTS ON A FIN STABILISED BOMB MODEL,

BY

N. POLLOCK, R.G. BROADBENT L.J. ROBERTS

(1) Jun 77

SUMMARY

Transonic wind tunnel tests have been conducted on a fin stabilised bomb model. The tests covered a Mach number range of 0.95 to 1.20 and a pitch angle range of -2° to 28° for roll angles between 0° and 45°. The model exhibits normal longitudinal stability characteristics except for pitch angles between 14° and 21°, roll angles greater than 15° and a Mach number of 0.95 where a small unstable region was observed. The lateral force and moments were small for pitch angles up to 4°. At higher pitch angles and unsymmetrical roll angles significant lateral force and moments occurred.

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ABSTRACT: Transonic wind tunnel tests have been conducted on a fin stabilised bomb. The tests covered a Mach number range of 0.95 to 1.20 and a pitc' range of -2° to 28° for roll angles between 0° and 45°. The model normal longitudinal stability characteristics except for pitch angle etween 14° and 21°, roll angles greater than 15° and a Mach number of 0.95 where a small unstable region was observed. The lateral force and moments were small for pitch angles up to 4°. At higher pitch angles and unsymmetrical roll angles significant lateral force and moments occurred.

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NOTATION

(Headings used in data listings are given in brackets)

Cx	(AXIAL)				Axial force coefficient Axial force/ $\frac{1}{2}\rho$ V ² S
CY	(SIDE F	r)		=	Side force coefficient side force/2 p V2S
c _N	(NORMAI	.)		=	Normal Force coefficient Normal force/2p V ² S
cı	(ROLL M	1)		=	Rolling moment coefficient Rolling moment/2p V2Sb
Cm (PITCH)			=	Pitching moment coefficient Pitching moment/2 p V2Sb
Cn	(YAW M)			=	Yawing moment coefficient Yawing moment/2 p V2Sb
Cxp	(BASE)			=	Base force coefficient (Pbase - Po) Sb/2 o V2S
CD	(DRAG)				Drag coefficient corrected for Po acting on base
	where	P V S b			Free stream density Free stream velocity Body cross section area 2160.6 mm ² Body diameter 52.45 mm
		Sb	=		Base area 595.5 mm ²
		Pbase	=		Pressure acting on base area
		Po	=		Free Stream static pressure
9 (7	HETA)				Pitch angle
Ø (F	LANG)				Roll angle
Mo (N	MACH)				Free stream Mach number
R (F	EYN)				Reynolds number based on b.
	ER)				Serial number
		The sign co	nve	nti	on for coefficients and angles i

- is
 - (ii) The moment reference centre is on the model centreline 124.6 mm aft of the nose.

1. Introduction

At the request of the Weapons Research and Development Wing of the Weapons Research Establishment a series of transonic wind tunnel tests were carried out on a model of a fin stabilised bomb. To avoid the construction of a new model for these tests, an existing model and strain gauge sting balance were supplied by W.R.E. This model had been tested at subsonic speeds at W.R.E. and the tests reported here were required to investigate the longitudinal and lateral stability of the bomb at transonic and low supersonic speeds.

The tests reported in this memo were conducted in the ARL transonic wind tunnel during December 1976.

Test Details

2.1 Model

A dimensioned drawing of the model is presented in Fig.2. Boundary layer transition was fixed on the body by two rows of 0.12mm diameter spheres located 20mm aft of the nose.

2.2 Wind Tunnel

These tests were conducted in the ARL transonic wind tunnel. The nominal dimensions of the tunnel test section are 813mm high by 533mm wide. For these tests the test section walls were all longitudinally slotted (Fig.3) with an open area ratio at the model location of 10.5%.

Mach number and dynamic pressure were derived from measurements of the static pressure in the plenum chamber surrounding the test section, and in the contraction entry, assuming these to be the static and total pressures of the test section flow respectively.

The frontal area of the model at zero incidence was 2252.2 mm²giving a blockage ratio of 0.52%. Because of this low blockage ratio no corrections for the effects of tunnel interference were applied.

2.3 Test Program

Six component force and moment coefficients were measured over a pitch angle range of $-2^{\circ} \le 0 \le 28^{\circ}$ in 2° increments. Roll angle was varied in the range $0 \le \emptyset \le 45^{\circ}$ in 7.5° increments. Due to the symmetry of the model the aerodynamic characteristics in the range $0 \le \emptyset \le 360^{\circ}$ can be determined from the tests in a single quadrant of \emptyset . The range of Mach number covered for each attitude was $0.95 \le M \le 1.20$ in increments of 0.05. In addition, several runs were conducted at zero roll angle and various pitch angles with fine increments in Mach number to investigate the effects of wall reflected wave interference.

The variation of test Reynolds number with Mach number is presented in Fig.4.

3. Results and Discussion

Tests in slotted wall transonic wind tunnels are subject to interference from waves generated by the model being reflected back onto the model at low supersonic speeds. This interference covers the Mach number range from just above 1.0 until the reflection on the bow shock moves downstream of the model base. For the tests reported here the Mach number range for wall reflection interference is estimated to be $1.00 \le M \le 1.11$ (Fig.5). The effect of this interference is illustrated in Fig.6. In this Figure Cm is plotted against M with small increments in test Nach number for $\theta = 80$, 14° , and 28° . The lines drawn through the experimental points are considered to be the probable interference free results.

A full listing of the results of the main test program (not including the fine M increment runs) is presented in Table 1. To illustrate the trends of the data some of the M = 0.95 and M = 1.20 results are plotted. These two Mach numbers were chosen so that the discussion would not be complicated by wall reflection interference. In Fig. 7 CN is plotted against 9 for $\emptyset = 0^{\circ}$, 22.5° and 45°. It can be seen that for pitch angles up to about 80 Ø has very little effect on CN. For larger pitch angles CN is reduced as \$\psi\$ is increased from 00 to 450. For CN up to about -1.0 ($\theta \approx 8^{\circ}$) \emptyset has little effect and the model is stable. For larger values of C_N the stability is reduced as \emptyset is increased from 0° to 45°. At M= 0.95 an unstable region appears at -1.5> $C_N > -2.5 (14^{\circ} < 9 < 21^{\circ})$ as the model is rolled away from $\emptyset = 0$? Although not plotted, the M = 1.00 and 1.05 results also show vestiges of this unstable region. In Fig. 9 C1 is plotted against θ for $\emptyset = 0^{\circ}$ to 45° in 7.5° increments. As would be expected for the two symmetrical configurations $\emptyset = 0^{\circ}$ and 45° , C_1 is near zero over the whole pitch angle range. For values of \$\psi\$ other than 0° and 45°, C1 is near zero up to 0 = 8°. For larger values of 0 an increasing positive C1 appears. The largest C1 values occur for $\emptyset = 22.5^{\circ}$.

In Fig. 10 Cy is plotted against θ for $\emptyset=0^\circ$ to 45° in 7.5 increments. A similar plot of C_n is presented in Fig.11. From Figs. 10 and 11 it can be seen that for the two symmetrical configurations $\emptyset=0^\circ$ and 45° , Cy and C_n are small throughout the test pitch angle range. For values of \emptyset other than 0° and 45° significant Cy and C_n values occur for $\theta>4^\circ$. The largest C_Y and C_n values occur for $\emptyset=15^\circ$.

4. Conclusion

Transonic wind tunnel test on a fin stabilised bomb model supplied by WRE have been conducted. The tests covered a Mach number range of $0.95 \le M \le 1.20$ in 0.05 increments, a pitch angle of $-2^{\circ} \le 9 \le 28^{\circ}$ in 2° increments and a roll angle range of $0^{\circ} \le \emptyset \le 45^{\circ}$ in 7.5° increments.

The model has positive longitudinal stability over most of the test range. However for $14^{\circ} < 9 < 21^{\circ}$, $\emptyset > 7.5^{\circ}$ and M = 0.95 there is a small unstable region. For roll angles other than 0° and 45° the model exhibits significant Cy and Cn when a $>4^{\circ}$ and significant C₁ when a $>6^{\circ}$ For $\emptyset = 0^{\circ}$ and 45° Cy, C_n and C₁ are small through the entire pitch angle range.

TABLE 1.1

DATA LISTINGS

	917 W		. 00013 . 0000 . 0000 . 0000 . 0000	. 6661 . 6661 . 6666 . 6666	.0000 .0000 .0000 .0001	.0002 .0001 .0004 .0011	. 6882 . 6889 . 6826 . 6828	.0001 .0010 .0000 .0000	. 0000 . 0000 . 0000 . 0000 . 0000
	YAW P	0.2082 0 0.0031 0 0.0131 0 0.0169 0	0.0550 0 0.0438 0 0.0171 0 0.0195 0	0.0074 0.0117 -0 0.00990 0 0.0064 0	0.0198 0 0.0147 0 0.0778 0 0.0192 0	0.0017 -8 0.0178 0 0.0294 0 0.0379 0	0.0135 9 9.0382 9 9.0363 9	0.8892 8 8.8845 -8 8.8819 -8 8.8176 8	3.2082 -8 9.2082 -8 9.2081 -8 0.2068 8 0.3251 8
	S10E F	0.0094 -0 0.0011 0.0011 0.0157 -0	0.0193 - 0 0.0145 - 0 0.0157 - 0 0.0155 - 0	0.000.00 0.000.00 0.000.00 0.000.00 0.000.00	0.2140 -0 0.3123 -0 0.0277 -0 0.0095 -0	0.0243 0.0132 - 0 0.0144 - 0 0.0278 - 0	0.0070 0.0015 0.0198 0.0156 0.0156	0.3055 - 0 0.3031 - 0 0.3022 0 0.3132 - 0	0.0226 0.0027 0.0027 0.0125 0.0125
	ORAG	0.2974 0.3263 0.5166 0.6848	0.5984 0.5824 0.5881 0.5985	0.3326 0.2886 0.2905 0.3217	0.5965 0.5912 0.6018 0.3149	0.5208 0.6130 0.6042 0.6174 0.3427	0.3728 0.5615 0.6649 0.6440	0.3892 0.4863 0.5981 0.7228	6.7113 6.5082 6.6759 6.7895
DEG.	BASE	-8.00066 -8.01145 -0.01146 -0.01148	-0.0592 -0.0542 -0.0542 -0.0220	-0.0131 -0.0128 -0.0128 -0.0153	-0.0208 -0.2474 -0.0612 -0.0172	8:0132 -0:0438 -0:0569 -0:0648	-0.0202 -0.0110 -0.0461 -0.0600	-0.0199 -0.0291 -0.0581 -0.0581	-0.0688 -0.0305 -0.0349 -0.0349 -0.043
10 11	AXIAL	22998 23355 24923 64923 6388	-0.6514 -0.6297 -0.6297 -0.6126	20 3458 -0 3014 -0 3305 -0 4896	-6.6111 -0.6307 -0.6568 -0.3073	-0.6388 -0.6388 -0.6319 -0.6564	0 1 2 3 3 3 1 9 9 9 9 9 9 9 9 9 9 9 9 9 9 9 9	-0.2964 -0.3217 -0.6688 -0.6688	0.06671 0.38898 0.38898 0.06584 7889
ROLL ANGLE	PITCH	0.1355 0.1483 0.8928 0.2587	0.1966 0.0464 0.0793 -0.0793	0.0022 0.0103 -0.1445 -0.1761	-0.1423 -0.2899 -0.1338 -0.3328	-0.2051 -0.3668 -0.4776 -7.3216	-0.5659 -0.3582 -0.6721 -0.7818	-0.9555 -0.8792 -0.5528 -0.9992 -0.9413	-0.8849 -1.19446 -0.7534 -1.3126
	NOMMAL	0.1472 2.1577 3.1496 3.2063	6.1881 8.6221 -8.6251 8.6318	0.0016 0.0042 -0.1752 -0.1952	-8.1891 -8.2492 -8.1882 -8.3642	-8.4027 -8.4378 -8.5884 -8.5884	-8.2987 -8.2533 -8.6463 -8.6544	-8.6254 -2.6337 -0.7734 -3.9018	-0.8483 -1.6771 -1.6942 -1.1725 -1.1133
	RANG	22022 23427	22220 22244	22229	22222	20202	20000	2000	20200 0 24402 4
	THETA	-2.62 -2.62 -2.63 -2.63	- 2.02 - 6.03 - 6.01 - 6.01	22.61 22.60 20.60	22.22.4 20.22.4 100.20.	4 4 4 4 6 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8	66.66 66.66	8 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8	8.86 18.03 18.05 18.09 18.05
	APOH	20.451 1.058 1.058	11111	110011 00000 00000 00011	1116 1116 1116 1116 1116 1116 1116 111	111.00 111.00 111.00 113.00 103.00	11.000 11.000 11.000 12.00 12.00	0 0 1 1 1 0 0 0 0 1 1 1 1 0 0 0 0 1 1 1 1 0 0 0 0 0 1 1 1 1 0	11.00 11.00 11.00 11.00 11.00 11.00 11.00
	REYN	0.198 0.198 0.198 0.198	0.296 0.296 0.294 0.199	0.191 0.186 0.173 0.176	0.1198 0.1198 0.1195 0.173	0.183 0.186 0.191 0.191	0.188 0.185 0.190 0.211 0.198	6.135 6.135 6.196 6.195	0.201 6.181 6.193 6.193 6.199
	SER	0 M 4 M 4	V 8 9 6 1 1	11111	2222	22 22 22 24 25 26 26	33333 335 335 335 335 335 335 335 335 3	8 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8	44444 4 HGW4R 0

TABLE 1.1 DATA LISTINGS

ROLL ANGLE . Ø DEG.

TABLE 1.1

DATA LISTINGS

	ROLL M	0.0041 0.0119 0.0028 0.0072		0.0038 0.0057 0.0022 0.0055
	Y AW M	0.0365 -0.0333 -0.0041 0.0351 -0.0562 -0.0119 0.0276 -0.0211 -0.0028 0.0457 -0.0052 -0.0072	0.0238 0.0317 0.0707 0.0604	7 -0.8333 -0.8838 1 0.8862 -0.8857 7 -0.8527 -0.8822 8 -0.8986 -0.8055
	SIDE F	0.0355 0.0351 0.0276 0.0457	0.0128 0.0238 0.0341 0.0341 0.0343 0.0367 0.0552 0.03694 0.03694	0.0291 0.0291 0.03447
	DRAG	2,0213 2,0272 1,9859 2,1445 2,3275	2.4080 2.3132 2.3160 2.2732 2.4598	2.6742 2.7246 2.6385 2.6226
£6.	BASE	-8.89876 -8.8987 -8.8599 -8.8522	-0.1019 -0.1015 -0.0971 -0.0732 -0.0547	-0.0539 -0.1145 -0.1047 -0.0976
ANGLE = 0 DEG	AXIAL	-3.7127 -2.7318 -0.6345 -0.8976 -3.6948 -2.6684 -0.6401 -0.8987 -4.8896 -2.9358 -0.2589 -0.8599 -4.2255 -2.8418 -0.3489 -0.8522 -4.2593 -2.7424 -0.5281 -0.8466	-0.1 -4.2959 -3.2386 -0.6573 -0.1019 -0.1 -4.1402 -2.9320 -0.6267 -0.1015 -0.1 -4.1259 -2.8897 -0.6311 -0.0971 -0.1 -4.4820 -3.0094 -0.2335 -0.0732 -0.1 -4.6388 -2.9497 -0.3378 -0.0547	0,1 -4,7469 -3,0237 -0,5195 -0,0539 -0,1 -4,7465 -3,4303 -0,6397 -0,1145 -0,1 -4,5632 -3,1772 -0,6099 -0,1047 -0,1 -4,5494 -3,0933 -0,6105 -0,0976
ROLL ANG	PITCH	-2.7318 -2.95684 -2.9358 -2.7424	-3.03386 -2.9328 -3.08597 -3.08994	-3.8237 -3.4383 -3.1772 -3.8933
	NORMAL	13.7107 13.6948 14.8896 14.8255	-4.1402 -4.1259 -4.4820 -4.6388	-4.7469 -4.5832 -4.5494
	RANG		90000	4444
	THETA	24 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2	26.46 26.43 28.45 28.45 28.46 28.46	28.58 28.47 28.49 28.51
	HACH	11.199 11.99 11.99 11.99 11.99	11.121	1.049
	REYN	8.286 8.286 8.178 8.185	0 1196 0 1196 0 1199 0 1181	8.195 1.849 8.195 1.188 8.199 1.149
	SER	99999	11000	1001

TABLE 1.2 DATA LISTINGS

ROLL ANGLE = 7.5 DEG.

ROLL M	00000 00000 00000 00000 00000	00000 00000 00000 00000 00000 100000 100000 100000 100000	00000 00000 00000 00000 00000 00000 0000	0 00000	-0.0000 -0.00010 -0.00013 0.00013	0.00023 0.00014 0.0019 0.0019 0.0060
YAW M	0.0015 0.0102 0.01103 0.00113	0.00042 0.00042 0.00030 0.00030 0.00030 0.00030 0.00040 0.00040	0 1 1 0 0 0 1 0 0 0 0 0 0 0 0 0 0 0 0 0	. 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0	-0.0967 -0.0967 -0.0265 -0.0601	. 0899 . 0899 . 0899 . 0898 . 0888 . 0883
SIDE F	-0.0027 -0.0027 -0.0065 0.0057 0.007	0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0	00.00 00		0.9358 0.0258 0.9126 0.9279	0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.0
DRAG	0.2945 0.3031 0.5021 0.5020	0 25843 0 72543 0 725543 0 75852 0 75852 0 75833 0 75833 0 75833	0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0	w www.	0.3755 0.4110 0.5943 0.7118	0.7043 0.4736 0.6537 0.7756
BASE	-0.0058 -0.0141 -0.0204 -0.0167	-0.00551 -0.00551 -0.00504 -0.0029 -0.0029 -0.0158 -0.0158	001010 00000 00000 00000 00000 00000 00000 0000	6 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0	-0.0270 -0.0212 0.0050 -0.0479	- 6 . 6 . 6 . 6 . 6 . 6 . 6 . 6 . 6 . 6
AXIAL	-0.2954 -0.3163 -0.5921 -0.5921	100 29471 100 29471 100 29472 100 5882 100 5981 100 5981 100 5981 100 5981 100 5981 100 5981	0 - 1 - 0 - 0 - 0 - 0 - 0 - 0 - 0 - 0 -	0,287 0,309 0,478 0,621 0,637	-0.2913 -0.3194 -0.6483 -0.6483	.0,6628 -0,2845 -0,3175 -0,4892 -0,6533
PITCH	0.1347 0.1493 0.2348 0.0717	0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0	201-1-0-1-0-1-0-0-1-0-1-0-1-0-1-0-1-0-1-	8.591 8.337 8.636 8.562 8.552	-0.9462 -0.8680 -0.5310 -0.9690	-0.7923 -1.2692 -1.1743 -0.7116 -1.2692
NORMAL	0.1465 0.1557 0.1488 0.1978	5 -1 - 1 - 1 - 1 - 1 - 1 - 1 - 1 - 1 - 1	201121 20112 201121 20112 20112 20112 20112 20112 20112 20112 20112 20112 20112 20112 20112 20112 20112 20112 2011	6 . 0 6 7 6 . 0 6 7 6 . 0 6 7 6 8 2 6 8 2 6 8 2 6 8 8 8 8 8 8 8 8 8 8	-0.8151 -0.7635 -0.6948 -0.8692	-0.8426 -1.0583 -1.08829 -1.1620 -1.1620
RANG	,,,,,	V4.4.4.V VV4.V4:	7,4,7,4,7,7,7	, ,,,,,	V4141VV	VV4.VV V
THETA	22.22		00044 4444 00000 0000 01010 0000	0 00000	88888 8969 84748	8.06 10.03 10.05 10.09 10.06
AACH	8.95 1.832 1.851 1.152	100000000000000000000000000000000000000	11122 1111 11122 2111 18212 21113 18212 1121	2 2 2 2 1 2	0 9 9 9 9 1 1 1 1 5 1 1 5 1 1 5 1 5 1 5 1	10011 00000 00000 11 00000 0000 0000 0
REY	0.173 0.188 0.181 0.188	00.1195 00.1195 00.1195 00.1196 00.1198 00.1198	00.1193 00.1193 00.1198 00.1190 00.1190 00.1195	1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	6.176 6.186 6.185 6.196	0.196 0.176 0.181 0.195 0.195
SER	119 119 119 119	22 25 26 27 27 28 33 33 34 35 35 35 36 37	88888 8844 84687 8884	4 44444	0 1 0 E	0.00000

TABLE 1.2

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	ROLL M	0.0037 0.0029 0.0003 0.0010	0.0018 0.0063 0.0101 0.00111	0.0101 0.0047 0.0083 0.0167	0.0080 0.0151 0.0078 0.0098	.0.00003 0.00000 0.00000 0.00000	0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.0	0.0428 0.0613 0.0613 0.0519	8.8558 8.8559 8.86695 8.8687 8.8722
	Y AW M	-0.1092 -0.2269 -0.1288 -0.0573	-0.0983 -0.1269 -0.2915 -0.1455	-0.1596 -0.1034 -0.1377 -0.3579	-0.1412 -0.2183 -0.2082 -0.2186	-0.3289 -0.2438 -0.3009 -0.3565	-0.5413 -0.5023 -0.3972 -0.5218	-0.5267 -0.4825 -0.6592 -0.4724	
	SIDE F	0.0385 0.0811 0.0448 0.0239	0.0454 0.1054 0.0534 0.0534	0.0569 0.0414 0.0411 0.1274 0.0782	0.0515 0.0766 0.0790 0.0734 0.1864	0.1268 0.0943 0.1059 0.1647	0.2081 0.1858 0.1528 0.1847	0.1848 0.2456 0.1857 0.2239	0.23 0.1148 0.23 0.23 0.23 0.23 0.23 0.23 0.23
	DRAG	0.7837 0.5215 0.5634 0.7458	0.8692 0.6159 0.6838 0.8644	1.0189 0.9735 0.9963 0.7405	1.0042 1.1662 1.1177 1.1366 0.9326	1.0444 1.3463 1.3463 1.3463	1.1500 1.3141 1.4293 1.58839	1.5465 1.3770 1.5542 1.5542	1.0462 1.0462 1.08462 1.9368 2.0715
. 530	BASE	-8.8727 -8.8351 -8.8351 -8.8698	-0.0809 -0.0809 -0.0463 -0.0448	-8.0836 -8.0829 -8.0783 -8.0490	-0.0220 -0.0765 -0.0849 -0.0852	-0.0524 -0.0270 -0.0840 -0.0873	-0.0595 -0.0487 -0.0287 -0.0919	0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.0	-0.0917 -0.0895 -0.0895 -0.0546 -0.0448
E . 7.5	AX I AL	-8.6731 -0.2808 -0.3234 -0.5035	-0.6640 -0.2881 -0.3372 -0.5208	-0.6940 -0.6727 -0.6888 -0.2879	-0,5381 -0,6911 -0,6752 -0,6888	-8.3618 -8.5246 -8.6996 -8.6993	-0,2787 -0,3597 -0,5289 -0,6841	-0.265 -0.365 -0.3569 -0.5292	0,6511 0,6511 0,2587 0,3586 0,5179
OLL ANGL	PITCH	-1.5919 -1.5919 -1.5626	-1.3326 -1.2846 -1.9026 -1.6081	-1.8258 -1.5456 -1.4914 -2.1619	-1.2323 -2.3293 -1.7580 -1.6934 -2.3531	-2.2682 -2.2682 -2.9588	-2.5145 -2.3327 -1.6484 -2.5067	-2.1396 -2.6780 -2.4813 -1.8601 -2.6888	12.3568 12.55689 12.5689 12.5689 12.8856
œ	NORMAL	-1.1037 -1.3066 -1.3459 -1.2488	-1.3690 -1.5712 -1.5733 -1.6429	-1.7518 -1.6457 -1.6553 -1.9462	-1.8510 -2.0712 -1.9760 -2.0006	-2.3976 -2.4884 -2.4139 -2.4692	-2.7393 -2.9562 -2.7687 -2.9727 -2.8518	-2.8541 -3.1406 -3.5565 -3.1996	-3.2338 -3.2536 -3.5534 -3.7361 -3.6451
	RANG	VV 4.4.V	*****	*****	*****	4.0.0.0	*****	*****	
	THETA	10.08 12.04 12.12 12.12	12.11 14.01 14.09 14.09	114 144 14.12 16.06 16.06	166.19 16.116 16.116 18.118	18.16 18.22 18.16 18.20 18.20	22.00 2.00 2.00 2.00 2.00 2.00 2.00 2.0	22.22 22.22 22.28 22.34	22.32 22.32 24.28 24.34 24.34 24.36
	4ACH	2000 2000 2000 2000 2000 2000	11.00 00.00 00.00 00.00 00.00 00.00	11.12 12.23 12.23 12.23 12.23 12.23 13.23	1.124 1.121 1.148 0.951	840 940 940 940 940 940 940 940 940	11.00.00	100 11 000 11 000 001 000 001	11.020.11
	REYN	0.198 0.181 0.185 0.196	0.199 0.203 0.181 0.195	00.196 00.201 00.201 1.803	0.193 0.201 0.201 0.206	0.176 0.180 0.185 0.185	6.176 6.176 6.186 6.185	0.195 0.170 0.176 0.183	0.191 0.195 0.195 0.195 0.193
	SER	44 8 9 9 9	27.78	54257	7 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8	88886	40000	900 000	1 11111

TABLE 1.2 Data Listings

	ROLL M	0.0719 0.0658 0.0812 0.073	0.6704 0.6738 0.0681 0.1065	0.1022 0.0756 0.0723
	YAH R	-0.6495 -0.6354 -0.8167 -0.8167	-0.6473 -0.665 -0.6678 -0.2612	-0.6252 -0.6389 -0.6889
	SIDE F	0.24 0.23 0.23 0.31 0.31 0.32 0.34 0.34 0.34 0.34	0.2378 0.2472 0.1333 0.1533	8.2572 8.2348 8.2486 8.2451
	DRAG	2.0841 2.0123 1.9196 2.0995 2.2679	2.3792 2.2926 2.2884 2.1358 2.3933	2.6986 2.6988 2.6984 2.5893
. 530	BASE	2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2	-0.1039 -0.0979 -0.1052 -0.1052	-0.0584 -0.1139 -0.1113
ANGLE = 7.5	AXIAL	7,6 -3,6619 -2,5886 -8,6486 -8,8984 7,6 -3,6591 -2,5867 -8,6461 -8,8964 7,6 -3,9574 -2,6364 -8,2657 -8,8844 7,6 -4,1469 -2,6428 -8,3447 -8,8688 7,6 -4,1356 -2,3878 -8,5228 -8,8478	7.6 -4.2463 -3.0645 -8,6543 -8,1074 7.6 -4.0964 -2,8156 -0,6273 -8,1039 7.6 -4.2697 -2.7157 -8,6282 -8,8979 7.5 -4.2254 -2,5268 -8,2471 -8,1052 7.6 -4.5325 -2,6844 -8,3357 -8,8716	7.6 -4.6141 -2.6381 -0.5141 -0.584 7.6 -4.6919 -3.2498 -0.6374 -0.1139 7.6 -4.5444 -3.0705 -0.6119 -0.1113 7.6 -4.4980 -2.9685 -0.6070 -0.1144
ROLL ANGL	PITCH	-2.5886 -2.6367 -2.6364 -2.36428	-3.00 -2.00 -2.7157 -2.5260 -2.6844	-2.6381 -3.2498 -3.0705 -2.9685
œ	NORMAL	-3.6519 -3.9574 -4.1469	44. 44. 44. 44. 44. 44. 44. 44. 44. 44.	4.4.4. 4.4.4. 4.4.4. 4.4.4. 4.4.4.
	RANG	010101010	01010101	7777
	THETA	24.28 26.38 26.38 26.38 26.38 26.38	25 2 2 2 2 2 2 2 2 3 2 3 2 3 3 3 3 3 3 3	28.52 28.52 28.58
	MACH	11.151	11.132	1.048 1.100 1.149
	REYN	0.195 0.198 0.178 0.185	0.195 0.199 0.201 0.178	8 1198 8 1195 8 1295
	SER	110	1115 1115 1116 1116	1119 128 121 122

TABLE 1.3 DATA LISTINGS

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	ROLL M	0.0002 0.0001 0.0001 0.0015	0.0015 0.0008 0.0010 0.0003	6.0019 6.0013 6.00013 6.0003	8.0013 8.0011 8.0007 8.0007	3.0006 9.0017 9.0017 9.0013	3.0003 3.0007 3.0036 3.0031	3.8817 3.8816 5.8816 5.8821	0.00035 0.00035 0.00036 0.0034	3.0024
	YAH M	0.0021 0.0021 0.0037 0.0018	0.0133 0.0114 0.0039 0.0389	0.0150 0.0355 0.0056 0.0039	0.0183 0.0183 0.0665 0.0203	0.0079 0.0382 0.0572 0.0855	0.00020 0.00317 0.0050 0.0098	0.1667 0.1248 0.0541 0.1144	0.0931 0.2681 0.1742 0.059	0,1224
	SIDE F	0.0070 0.0049 -0.0069 0.0079	0.0289 - 0.0019 -0.0030 0.0030	0.0122 - 0.0125 - 0.0007 - 0.0007 -	0.0129 0.0144 0.0222 0.0114	0.0068 0.0226 0.02240 0.02240	0.0235 0.02148 0.0258 1.0238 1.0314	0.0596 0.0278 0.0278 0.0462	0.0361 0.0973 0.08796 0.08786	0.0535 -
	DRAG	0.2871 0.5871 0.5896 0.5893	6.5927 6.2797 6.3165 6.4891	6.59 6.59 6.79 6.79 6.79 6.79 6.79 6.79 6.79 6.7	6.5833 6.5893 6.3871 8.3132	0.5140 0.6052 0.6079 0.6118	888.99 888.99 888.99 888.99 888.99 888.99 888.99 888.99 888.99	0.3885 0.5947 0.5947 0.6944	0.7081 0.4317 0.4758 0.6610	0.7732
. 530	BASE	-0.0000 -0.00150 -0.0181 -0.0232	-0.0624 -0.0158 -0.0158 -0.0169	-0.0402 -0.0528 -0.0154 -0.0156	0 - 0 - 0 - 0 - 0 - 0 - 0 - 0 - 0 - 0 -	0.0014 0.00333 0.00542 0.00542 0.00542	-0.0172 -0.0111 -0.0348 -0.0566	-0.00153 -0.00254 -0.0026 -0.0026	-0.0728 -0.0289 -0.0010	-0.0698
LE = 15	AXIAL	-0.2984 -0.4765 -0.6889	-0.6492 -0.2999 -0.3264 -0.4722	-0,6337 -0,6443 -0,3026 -0,3179	-8.5921 -8.6433 -8.2963	-0.4761 -0.6121 -0.6335 -0.6500	-0.3111 -0.4745 -0.6164 -0.6464	-0.2866 -0.3178 -0.4917 -0.6565	-0,6686 -0,2831 -0,3245 -0,4962	-0,6598
ROLL ANGI	PITCH	0.1319 0.1319 0.0969 0.2413	0.1791 -0.0077 -0.0177 -0.0247	-0.0655 0.0080 -0.1472 -0.1809	-0.1265 -0.2512 -0.1395 -0.3264	-0.1943 -0.3448 -0.3287 -0.5733	-0.5538 -0.3305 -0.6150 -0.5378	-0.919 -0.8169 -0.4871 -0.9425 -0.8857	-0.7698 -1.2072 -1.0788 -0.6444	-1.0747
	NORMAL	0.1418 0.1519 0.1428 0.1956	-0.0155 -0.0155 -0.0210 -0.0226	-8.0428 -0.0059 -0.1814 -0.1972	-8.1881 -8.2332 -8.1897 -8.3688	-0.3529 -0.3968 -0.4311 -0.3982 -0.5632	-8.5818 -8.5418 -8.6388 -8.6371	- 6 . 8 . 8 . 8 . 9 . 9 . 9 . 9 . 9 . 9 . 9	-8.8427 -1.8488 -1.8688 -8.9786 -1.1493	-1.0996
	RANG	44400 22223	2 2 2 2 2	00044 20022	00000 00000 0000 0000	00000 2 2 2 2 2	00000 00000 00000	04400 22223	22320	15.0
	THETA	22.62	-2 6 6 6 6 1 6 6 6 1 6 6 1 1 1 1 1 1 1 1 1	22.2.60	22.2.4.4 20.0.0.0 21.0.0.0 22.0.0.0	44440 99999 48888	00000 00000 00000	88888 8989 84788	1100 100 100 100 100 100 100 100 100 10	18.97
	HACH	8 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8	100001	1.158 1.199 1.999 1.051	1.149	11119 949 1159 1159 1159 1159	0 11 10 10 10 10 10 10 10 10 10 10 10 10	11.056	10.00.11	1,148
	R REYN	2 8 1176 5 8 1188 5 8 1188 5 8 1198 6 1198	7 0.196 8 0.176 9 0.181 0 0.188 1 0.193	2 0.196 3 0.199 4 0.178 5 0.185 6 0.188	17 8.193 18 8.196 19 8.281 22 8.173 23 8.176	4 6.186 6 6.196 7 6.196 8 6.173	9 6.186 6 6.183 1 6.188 2 6.211 3 6.214	37 Ø.168 38 Ø.175 39 Ø.198 40 Ø.201 41 Ø.185	42 0.190 43 0.170 44 0.173 45 0.180 46 0.185	47 0.190
	SE			наана	44400	*****	MMMM	www44	*****	•

TABLE 1.3 DATA LISTINGS

ROLL ANGLE = 15 DEG.

ROLL M	6.6681 6.6681 6.6681 6.119	99999 999	00000000000000000000000000000000000000	100000 00 10400 004	. 0885 . 0885 . 0885 . 115	00000000000000000000000000000000000000	0.1153
T MY	-0.1231 -0.3591 -0.1836 -0.0962	82.1.62 9.152 9.153 9.133 9.298 9.298	0 - 1 - 1 - 1 - 1 - 1 - 1 - 1 - 1 - 1 -	24.00. 20.00. 20.00. 20.00. 20.00. 20.00.	7.08 8.00 4.00	11.00 10.00 11.00 10	-0.9083
SIDE F	0.0442 0.1330 0.0692 0.0363	00100 00 00004 00	6 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0	011100 BE	321	0.08827 0.08827 0.03842 0.0386 0.0398 0.0398 0.0398 0.0398 0.0398 0.0398	0.3129
DRAG	0.5045 0.5780 0.7481 0.8895	8 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8	000 000 000 000 000 000 000 000 000 00	2000 2000 2000 2000 2000 2000 2000 200	88. 88. 88. 718.	1.6003 1.70003 1.7779 1.7779 1.7760 1.7760 1.7760 1.9760	2,0338
 BASE		99.99.99.99.99.99.99.99.99.99.99.99.99.	2	22222 22 4 0 0 0 0 0 4	000 000 000 000 000 000 000		-0.1085
AXIAL	.0 .0 .0 .0 .0 .0 .0 .0 .0 .0 .0 .0 .0 .	000 M W 000 000 000 000 000 000 000 000	0000000	357 691 601 601 601 601 601 601 601 601 601 60	533 663 675 2875	0 1 1 0 1 0 1 0 1 0 1 0 1 0 1 0 1 0 1 0	-0,6653
PITCH	-1.0064 -1.4995 -1.2544 -0.7987	827.49 827.49 82.49 82.49	400 800111 400 800111 400 8000 800 8000 800 8000 800 8000 800 8000 800 8000 800 8000	1.171 1.171 1.171 1.852 1.752 1.75 1.75 1.75	1.985 1.985 2.913 2.913	2. 1. 1. 2. 2. 2. 2. 2. 2. 2. 2. 2. 2. 2. 2. 2.	-2,5289
NORMAL	-1.8965 -1.2764 -1.3189 -1.2311 -1.4489	363 533 662 731 731	-1.0494 -1.0494 -1.7908 -1.9583 -1.9719	2.1572.1573.25.393.25.393.25.393.25.393.25.393.25.393.35.35.35.35.35.35.35.35.35.35.35.35.35	2.043 2.043 2.043 2.043	1.3. 1.3. 1.3. 1.3. 1.3. 1.3. 1.3. 1.3.	-3.6986
RANG	นนนนน จัจจัจจั		444 4444 446 6666 446 6666	-	ממה ממ	200 2000 200 2000 200 2000	15,1
THETA	12.08 12.08 12.08 12.08 12.08	94944 444	16.11 16.11 16.11 16.11 16.11 16.11 16.11 16.11 16.11 16.11 16.11	44444	www ww	22.35 22.35 22.29 22.35 24.29 24.29 24.29 24.29	24.37
MACH	11.098	40000 44	100 11110 100 011110 100 001110 100 0001110	001110 00 00040 00	B NO.	110011	1.898
REYN	81191		6 6 6 7 7 8 6 6 6 7 7 8 6 6 7 7 8 6 6 7 7 8 8 6 7 7 8 8 8 7 7 8 8 8 8	19999	244 44	88 88 88 88 88 88 88 88 88 88 88 88 88	8.185
SER	44666	M41-00 041	000 00000 0004 800000		30 1 E	489 69111 480 69111	113

TABLE 1.3 DATA LISTINGS

	ROLL M	8.1145	.1424	.1375	.1265	.1238	.1134	1.1130	.1698	.1644	.1563	11311	.1219	2.1119
	YAW M	-0.9480					•	•	G	<u>~</u>	_	_		-0.8392
	SIDE F	0.3355	0.1411 -	0.4084	0.2956 -	0.3022 -0.8708	0.3138 -	0.3186 -	0.1580 -	0.3719 -	0.2833	0.2877 -	0.2941 -	0.2957 -
	DRAG	1.9713				2,3178	2.2410	2.2441	2.0977	2,2984				2,5482
05.50	BASE	8 -2.3336 -8,6455 -8.1832	-0.0958	-0.0851	-0.0638	-8.1171	-0.1114	-0.1060	-0.1224	-0.0981	-0,5231 -0.0772	-0.1320	-0.1221	-0.1109
ANGLE = 12	AXIAL	-0.6455	-0,2783	-0,3719	-0,5308	-4.1133 -2.6996 -0.6612 -0.1171	-0,6315	-0,6364	-0,2642	-0,3686	-0,5231	-0,6500	-0,6218	-0,6233
MULL AND	PITCH	-2.3336	-2.2697	-2,0564	-1.8061	-2.6996	-2.5422	-2,4781	-2,2280	-2.0805	-1,9206	-2,8352	-2,7289	-2.6945
	NORMAL	-3.5798	-3.8841	-3.9452	-3.9199	-4.1133	-3.9872	-3.9709	-4.1442	-4.3194	-4.3501	-4.5396	-4.4152	-4.3934
	RANG	15.1	15,1	15.1	15,1			15,1			15,2	15,1	15,1	15.1
	THETA	24.38	26.36	26.43	26.48	26.43	26.44	26.46	28.42	28.49	28.54	28.50	28.51	28.52
	MACH	1.149	8,952	1.001	1.052	1.899	1,152	1.199	9.952	666.0	1.849	1.100	1.152	1.202
	REYN	6.198	8.175	0.181	0.185	8.198					8.185			
	SER	111	118	119	120	121	122	123	124	125	126	127	128	129

TABLE 1.4

		Y AW M	0.2084 0.3115 0.312 0.3145	-0.0530 -0.0008 0.0060 0.0327	-0.2138 -0.3231 0.3039 2.3086 0.3284	-0.3864 -0.8546 -0.8546 -0.8224	6.3814 -0.3334 -0.2383 -0.3876 -0.3771	-0.8577 -0.8529 -0.8529 -0.3782 -0.3992	-0.1678 -7.1481 -0.7553 -0.1871 -7.6995	-0.1292 -0.2127 -0.3979 -0.1532
		SIDE F	0.3036 -0.3031 -0.3051 -0.0068	6.3212 6.3847 6.3018 -6.3349 6.653	0.0145 0.01145 0.00104 -0.0010	6.0112 6.0122 6.0224 6.0024 6.0079	0.0237 0.0237 0.0196 0.0258	6.3217 6.3136 6.3244 6.3283	0.5594 0.0598 0.0598 0.0598 0.0598	0.2511 0.2556 0.2551 0.3545
		DRAG	0.3127 0.4992 0.5880 0.5958	6.5981 6.2899 6.3117 6.5802 6.5907	0.5888 0.5874 0.3864 0.3122 0.4890	0.5938 0.5876 0.5964 0.3835 0.3184	8.5152 8.6896 8.6839 8.6896	6.3617 6.5482 6.6423 6.6488	0.3571 0.3990 0.5845 0.7857 0.6914	0.7869 0.4216 0.6536 0.7835
	DEG.	BASE	-6.8123 -0.8116 0.8179 -0.8171 -5.8474	-0.0554 -0.0114 -0.0114 -0.0196	-0.0442 -0.0582 -0.3133 0.0097	-0.0130 -5.0130 -0.0567 -0.0146 -0.0126	6.8113 -0.8268 -0.8596 -0.8645 -0.3645	-0.0168 -0.0049 -0.0041 -0.00589	-0.0362 -0.0218 0.0829 -0.0459	-9.8695 -9.8271 -9.8271 -9.3653 -9.3653
LISTINGS	E = 22,5	AXIAL	-0.2939 -0.3193 -0.4767 -0.5987 -0.6387	-0.5475 -0.3031 -0.3231 -9.4807	-0.6330 -0.6556 -0.3011 -0.3189 -0.4729	-0.5998 -0.6385 -0.6468 -0.2941 -0.3133	-0.4809 -0.6103 -0.6352 -0.6481	-0,3288 -0,4819 -0,6141 -0,6355	-0.2872 -0.3125 -0.4827 -0.6355	-8 . 25 8 3 . 29 8 3 . 65 5 2 . 65 7 2
DATA	TOPE ANGLE	PITCH	0.1292 0.0979 0.2345 0.12345	0.1747 -0.3056 -0.3189 -0.3289	-0.3855 -0.1489 -0.11884 -0.1394	-0.1333 -0.2683 -0.1473 -0.3298	-0.1824 -0.3347 -0.4169 -0.3208	-0.5168 -0.2984 -0.5914 -0.0296	-0.8444 -0.7725 -0.8869 -0.8386	-0.7567 -1.1217 -1.0546 -0.5856 -1.1672
	œ	NORMAL	8.1427 2.1521 0.1432 8.1956 8.1429	6.1797 -3.0146 -6.0232 -2.0214 3.0842	-0.0465 -0.0368 -0.1784 -0.1963	-2.1893 -0.2362 -0.1986 -0.5533	-0.5488 -7.5922 -0.4231 -0.3937	-6.5319 -8.6226 -6.6338	-0.771 -0.7941 -0.7399 -6.6795 -0.6595	-1.08745 -1.5387 -4.9576 -1.1313
		RANG	22222 4.0.0.5 4.0.0.0	22222	22222	22222	2222 2222 4.0.025	22222 22222 4477	22222	222222 2
		THETA	22.22.2	-2 6.63 6.69 6.69 6.69 1.60	20.01 20.00 20.00 10.00 10.00 10.00	2.31 2.60 2.61 2.62 2.62	44446 6666 48866 48886	00000 0000 00000 00000 00000	88.89 89.69 86.65	8.06 10.03 10.03 10.03 10.05
		AACH	0.952 1.030 1.051 1.299	1.288 0.931 1.031 1.051	11.238 1.958 1.888 1.888	1.130 1.150 1.230 0.950	1.132 1.132 1.231 9.949	2.499 1.251 1.698 1.151 1.238	0.951 1.035 1.251 1.131 1.148	1.236 1.995 1.051 1.099
		REYN	6.178 6.193 6.193 6.193	0.241 0.181 0.185 0.195	0.199 0.233 0.181 0.193	6.198 6.198 6.233 6.178	8 1191 8 1191 8 1195 8 1198	6.196 6.196 6.193 6.193	0.178 0.198 0.198 0.195	0.291 0.173 0.176 0.181 0.193
		SER	W4 10 0 V	86948	844 644 74 74	22222	2288	33333	36 37 38 39 42	44444 4 447.67 Ø

TABLE 1.4

DATA LISTINGS

ROLL ANGLE = 22,5 DEG.

	ROLL M	6.0104 6.0073 6.0027 6.0062	22020 2	6.0264 6.0264 6.0271 6.0136	0.0264 0.0402 0.0334 0.0352 0.0611	6 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5	2693 2691 2691 2691	6.11028 6.11028 6.11632 6.11489 7.1397
	Y AW M	-0.1329 -0.3772 -0.1963 -0.1204	44444	-0.2826 -0.2685 -0.5897	-0.3048 -0.4948 -0.4497 -0.4381	-0.5395 -0.5395 -0.7212 -0.7907 -0.8216 -1.08496	11.007 9.00 9.00 10.00 1	0.9553 10.9553 10.9631 10.9631 10.9131
	SIDE F	0.0467 0.0315 0.0398 0.0398	250 250 250 250 250 250 250	0.1014 0.0030 0.2035 0.1250	0.1586 0.1586 0.1587 0.1393	0.2337 0.2339 0.2939 0.2939 0.2939 0.3939 0.3939 0.3939 0.3939	3.5 3.5 3.5 3.5 3.5 3.5 3.5 3.5 3.5 3.5	0.3351 0.3971 0.3188 0.1835
	DRAG	0.7827 0.4994 0.5560 0.7448	865 873 681 851	0.9682 0.9874 0.7889	0.9716 1.1364 1.1844 1.1249 0.8541	1.2952 1.2952 1.2952 1.2952 1.322 1.3163	410 410 410 410 410 410	1.6796 1.5957 1.5281 1.6424 1.8135
DEG.	BASE	-0.0720 -0.0290 -0.0317 -0.0042	0.037	-0.0833 -0.0814 -0.8493 -0.8493	-8.0161 -0.0881 -7.0827 -2.0894 -9.0477	22622 926 22622 926 22622 926 228622 926 21888 928 201848 44	6 6 6 6 6 6 6 6 6 6 6 6 6 6 6 6 6 6 6	-6.8928 -0.3928 -0.3968 -0.0643 -0.1120
E = 22,5	AXIAL	-0.6758 -0.2758 -0.3284 -0.5070	0,673 0,283 0,349 0,520	-0.6746 -0.6911 -0.2863 -0.3437	-0.5214 -0.6924 -0.6781 -0.6976 -0.2821	- 6.3536 - 6.5298 - 6.6813 - 6.6813 - 6.6879 - 6.2987 - 6.3713	676 658 368 543 676	-0,6496 -0,6684 -0,2969 -0,3886 -0,5733
ROLL ANGL	PITCH	-0.9580 -1.3779 -1.1239 -0.7157 -1.4881	217 166 625 241 828	-1.3644 -1.3326 -1.7966 -1.3841	-0.8931 -1.7827 -1.5188 -1.4517 -1.7596	1. 5932 1. 4248 1. 5213 1. 58213 1. 5932 1. 4932	999 739 668 651 548 142	-1.8739 -1.6207 -1.7947 -1.4353 -1.6828
	NORMAL	-1.2785 -1.2788 -1.2788 -1.2858	1.346 11.346 11.946 11.945 11.471	-1.6178 -1.5216 -1.7372 -1.8379	-1.7429 -1.9928 -1.9122 -1.9395	-2.1837 -2.5149 -2.2781 -2.52781 -2.5658 -2.5658	2.793 2.696 2.696 3.617 3.917 3.193	. 3. 3596 -3. 2483 -3. 5284 -3. 5284 -3. 5384
	RANG	22222	20000	22222	22222	20000 200	NN NNNN	0 00000 0 00000 0 00000
	THETA	10.08 12.05 12.08 12.12 12.12	11.0.1.1 11.0.1.1	144.13	16.28 16.15 16.17 16.19 18.12	18.19 18.24 18.26 18.22 18.22 20.27	22.23.32.23	22.34 22.37 24.32 24.39 24.45
	YN MACH	95 1,199 76 8,951 76 8,999 88 1,852 85 1,899	11011	91 1.151 95 1.231 76 8.958 88 1.888	85 1.051 88 1.150 93 1.150 96 1.232 70 0.950	98 1. 658 8. 1. 658 8. 1. 658 8. 1. 1. 1. 1. 1. 1. 1. 1. 1. 1. 1. 1. 1.	11 12 12 12 12 12 12 12 12 12 12 12 12 1	6 1.151 6 1.272 6 9 9 8 6 1.186
	SER RE	94 CN 4 R	00000	113	16 0.18 17 0.18 19 0.11 22 0.11	22.55	00 00000	38 6.13 39 6.13 42 6.13 44 6.13 45 6.13

TABLE 1.4 DATA LISTINGS

	ROLL M	8.1160 8.1398 8.1728 8.1731	0.1396 0.1280 0.1145 0.1928	2.1788 0.1481 0.1368 0.1279
	YAK	0.3186 -0.9233 0.3450 -0.9090 0.1085 -0.2845 0.3191 -0.8903 0.1455 -0.3832	0.3078 -0.3731 0.3014 -0.8648 0.2927 -0.3668 0.1442 -0.3769 0.3076 -0.8572	2,3723 0,1321 -0,3189 2,5852 0,2766 -0,7865 2,5120 0,2837 -0,7806 2,5001 0,2734 -0,7873
	SIDE F	0.3188 0.3858 0.1885 0.3191 0.1455	0.3078 0.3014 0.2927 0.1442	0.1321 0.2756 0.2887 0.2887
	DRAG	1.9291 1.9394 1.8148 1.9226 2.0762	2.2746 2.2153 2.2199 2.0502 2.2219	2,3723 2,5852 2,5120 2,5861
DEG.	BASE	-6.1047 -6.1029 -6.1107 -6.1016 -6.0664	22,7 -4.4124 -2,3373 -8,6737 -8,1298 22,5 -3.4128 -2,2412 -8,6434 -8,1166 22,5 -3.4988 -2,1776 -8,6437 -8,1863 22,7 -4.4298 -1,8980 -8,2627 -8,1143 22,7 -4.1257 -1,4425 -8,3889 -8,1829	-0.0395 -0.1379 -0.1226 -0.1167
ROLL ANGLE = 22,5 DEG.	AXIAL	-3.48/11 -2.053/ -0.6445 -0.1047 -3.48/7 -1.9892 -0.6524 -0.1029 -5.7276 -1.9229 -0.2892 -0.11/7 -3.7251 -1.3964 -0.3947 -0.1016 -3.5983 -1.0800 -0.5607 -0.0664	-0.6737 -0.6434 -0.6437 -0.2627 -0.3889	22.7 -4.1218 -1.2131 -0.5443 -0.0895 22.7 -4.4467 -2.4987 -0.6633 -0.1379 22.7 -4.3262 -2.3957 -0.6294 -0.1226 22.7 -4.3037 -2.3620 -0.5308 -0.1167
OLL ANGL	PITCH	-2.0537 -1.9892 -1.9229 -1.3964 -1.0800	-2.3373 -2.2412 -2.1776 -1.8980	-1.2131 -2.4987 -2.3957 -2.3628
ď	YOMMAL	-3.48/11 -3.48/17 -3.72/6 -3.72/51	-4.0124 -3.4128 -5.8988 -4.0298	-4.1218 -4.4467 -4.3262 -4.5037
	RANG	22.0 22.0 22.1 22.7	22.7	22.7 22.7 22.7
	THETA	24.42 26.39 26.39 26.55	26.46 26.47 26.49 26.49 26.53	26.52 26.52 26.53
	AACH	1.158 0.952 1.038 1.038	11.899 11.230 8951 8951	1,252
	REY	0.193 1.158 0.196 1.198 0.175 0.952 0.181 1.000	8,193 1,899 6,195 1,146 8,196 1,236 6,175 8,951 8,181 8,998	8.195 1,852 8.195 1,130 8.195 1,150 8.199 1,198
	SER	4444 0 0 C C O O	125 125 125 125 125 125 125 125 125 125	56 57 58 59

TABLE 1.5 DATA LISTINGS

	ROLL M	0.0014 6.0011 6.0011 6.0017	6.60 6.60 6.60 6.60 6.60 6.60 6.60 6.60	6.2226 6.2226 6.2228 6.2228	2.0015 0.0015 0.0012 0.0013	0.0000 0.0000 0.0000 0.0016 0.0018	3.0000 3.0000 3.0001 6.0016 7.0031	2.2011 2.2083 2.3083 2.3083	3.9073 9.9006 9.7003 3.70067 0.8032
	YAW	-0.3098 0.3270 0.0270 0.0058	-0.0607 0.0026 -9.0033 0.0184 0.0031	-0.0166 -0.0331 -0.0003 0.0010	-0.0043 -9.0112 -0.0605 -0.3187	-0.0088 -0.0326 -0.3403 -0.3823	-8.3532 -8.3639 -8.3588 -8.3588	-0.1423 -0.1112 -0.0591 -0.1034	-0.11448 -0.1559 -0.0677 -0.1118
	SIDE F	0.0031 -0.0038 -0.0051 0.0059	0.0014 0.0036 0.0024 0.0043 4.00.0	0.0097 0.0097 0.0020 0.0016 -0.0028	0.0111 0.01111 0.0234 0.0041	0.0014 0.0162 0.0158 0.3244	6.3198 6.3295 6.3295 6.3237	0.7494 0.7354 0.7217 0.9385 0.9389	0.35493 0.3218 0.3218 0.3218 0.3353
	DRAG	0.3888 0.3154 0.4959 0.5916	0.5987 0.2837 0.3162 0.4913	0.5754 0.5754 0.2795 0.3828 0.4930	0.5871 0.5911 0.2975 0.3184	6.5869 0.6829 0.6879 0.6189	6.3463 6.5416 6.6498 6.6434 6.6434	0.3626 0.4030 0.5873 0.6917	0.78445 0.54445 0.54445 0.7683
• 530	BASE	-0.0079 -0.0064 -0.0134 -0.0134	-0.2628 -0.2121 -0.2693 -0.2166 -0.665	-0.0413 -0.0605 -0.0163 -0.0143	-0.0176 -0.0514 -0.0589 -0.0127	0.0103 -0.0207 -0.0538 -0.0628 -0.0183	-0.0194 0.0091 -0.0283 -0.0569	-0.0191 -0.0213 0.0868 -0.0537 -0.0598	-3.8221 -5.8224 -5.8274 -6.9698 -3.9691
ANGLE = 30	AXIAL	-0.2919 -0.3167 -0.4712 -0.5985	-0.6455 -0.2958 -0.3255 -0.4747	-0.6332 -0.6368 -0.2899 -0.3186	-0.5985 -0.6386 -0.2866 -0.3832	-0.4736 -0.5979 -0.6341 -0.6477	-8,3898 -6,4882 -7,6166 -1,6381 -3,6579	-0,2792 -0,3156 -0,4842 -6,6389	-0,6647 -0,3017 -0,4892 -0,6530
ROLL ANG	PITCH	8.11382 8.1847 8.2378 8.6868	0.1777 -0.0030 -0.0127 -0.0163	-0.0271 0.0042 -0.1458 -0.1761 -0.1327	-8.1286 -9.2313 -8.1525 -8.2962 -0.3889	-0.1758 -0.3857 -0.3871 -0.3288	-0.4843 -0.2726 -0.5741 -0.5885 -0.5259	-0.7818 -0.7137 -0.4137 -0.8433	-8.7266 -1.9345 -9.451 -5.5428 -1.0894
	VORMAL	0.1432 0.1587 0.1457 0.1962	-0.0122 -0.0122 -0.0183 -0.0230 -0.0230	-8.0285 -8.0068 -8.1756 -6.1928	-6.1871 -6.2268 -6.1917 -7.5451 -8.5624	-8.3858 -8.3858 -6.4139 -8.3934	-4.5558 -3.5216 -4.6151 -4.6217	-8.7522 -8.7751 -8.7262 -8.8581 -8.8332	-2.4236 -2.4756 -3.4931 -1.1074
	RANG	33233	88888 8888 8888 8888 8888 8888 8888 8888	23.00	388333333333333333333333333333333333333	3883388	33333	326.22	8 88485 8 84485 8 84485
	THETA	-2.01 -2.01 -2.01 -2.01	00000	2.99 2.91 2.91 2.91	2.91 10.20 10.00 1	4444 8006 8006	66.00 66.00 86.00 84.00 84.00	8 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8	6.86 10.84 16.86 16.13 18.66
	MACH	0 0 0 0 1 1 1 0 0 0 0 0 0 0 0 0 0 0 0 0	1.198 1.661 1.661 1.32	1.151	1,121 1,148 1,238 0,951 0,951	1.648 1.138 1.149 1.199	0.999 1.051 1.058 1.150 1.230	6.951 1.699 1.698 1.151	1.238 8.958 8.999 1.858 1.298
	REYN	6.181 6.198 6.198 6.195	6.181 6.198 6.198 6.193	8.281 8.286 8.181 8.193	0.199 0.204 0.206 0.173	6.183 6.193 6.193 6.195	61198 61193 61193 61191	6.175 6.185 6.185 6.198	8.198 8.178 8.176 8.193 8.198
	SER	0 W 4 W 0	110007	21112	23222	460000	3321	3 9 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8	04444 4 00W40 0

TABLE 1.5 DATA LISTINGS

	ROLL M	0.0044 -0.00044 0.0054 0.2054	0.0075 0.0162 0.0135 0.0043	2.022 2.022 2.022 2.022 2.022 2.022 2.022 2.022 2.022 2.022 2.022	6.6333 6.0414 6.0308 6.0310	6.0238 6.02483 6.0575 6.0388	6.1177 9.6552 9.8552 6.8820 9.9745	0.0668 0.1355 0.1000 0.1213	0.0842 0.1513 0.1405 0.1357
	YAM	-0.1434 -0.2889 -0.1517 -0.1086	-0.1595 -0.2079 -0.3834 -0.1988	-0.2793 -0.2722 -0.2777 -0.4923	-3.2984 -0.4621 -3.4318 -3.4382 -6.6103	-2.5985 -0.4715 -8.6395 -9.7267 -8.7186	-7.8407 -7.8407 -7.5241 -7.8396 -0.3148	-0.7388 -0.2979 -0.6819 -2.1438	0.7712 0.7496 0.2368 0.5332 0.5332
	SIDE F	0.2413 0.1025 0.0499 0.3369	0.0626 0.0656 0.1259 0.0640	0.0936 0.1918 0.1934 0.1638	0.1613 0.1538 0.1548 0.1435	0.2153 0.1652 0.2141 0.2760	0.1259 0.2888 0.1864 0.2853	0.2498 0.1113 0.2421 0.2597	0.25556 0.1910 0.25556 0.1910 0.25556
	DRAG	0.7748 0.4912 0.5424 0.7342	2.8536 2.8648 2.5844 2.6462	8.9773 8.9538 8.9663 8.7812	0.9542 1.1073 1.0777 1.0992 0.8352	0.9238 1.8994 1.2551 1.2359 1.2658	0,9922 1,1619 1,3051 1,4866 1,4282	1.4425 1.1979 1.3488 1.4963	1.6679 1.6629 1.5834 1.7254 1.9548
. 530	BASE	-0.0803 -0.0274 -0.0291 -0.0011	-0.0747 -0.0767 -0.0338 -0.0319	-0.0778 -0.0821 -0.0838 -0.0356	-0.0100 -0.0844 -0.0803 -0.2801	-0.2433 -0.0135 -0.0834 -0.0827	-0.0816 -0.0543 -0.0541 -0.0941	-0.0854 -0.0944 -0.0751 -0.0662 -0.0960	-0.0922 -0.0926 -0.0916 -0.0916 -0.0817
30	AXIAL	0.6771 0.2747 0.3187 0.4986	6658 6785 6785 63255 63255	-8.6725 -8.6728 -8.6859 -8.2816 -8.363	-2,5145 -6,6795 -6,6861 -6,6861	-8.3362 -8.5288 -8.6721 -8.6686	-0.3889 -0.5292 -0.6888	-0.3833 -0.3833 -0.3717 -0.5618	0.6487 0.6587 0.2916 0.3905 0.5589
ROLL ANGLI	PITCH	-0.9238 -1.2732 -1.0546 -6483	-1.1465 -1.1125 -1.4826 -1.1136	-1.2799 -1.2799 -1.5959 -1.5959	-2.7276 -1.5993 -1.3433 -1.2636	-1.1336 -1.6349 -1.3723	-1.2634 -1.1381 -0.6156 -1.7331 -1.4564	-1.4343 -1.3000 -0.9937 -0.5019	11-156874 11-56874 11-56874 11-59874 11-59874 11-5988
	NORMAL	-1.0693 -1.1943 -1.2388 -1.1798	-1.3178 -1.4251 -1.4813	-1.6417 -1.2676 -1.2639 -1.6778 -1.7498	-1.6828 -1.9249 -1.6324 -1.0548	-2.2655 -1.9723 -2.286 -2.1968	-2.2752 -2.5340 -2.4052 -2.7042 -2.5915	-2.6835 -2.8256 -2.7246 -3.6936	-2.9877 -3.0838 -3.1716 -5.1154
	RANC	22723	8 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8	88888 88888 88888	8 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8	22222 2322 2322 2322 2322 2322 2322 23	23 23 25 24 44 4	3388	333333
	THETA	12.08 12.09 12.69 12.13	12.10 12.11 14.07 14.12	14.13 14.13 16.89 16.16	16.21 16.16 16.18 16.23 18.14	18.26 18.25 16.19 18.23	26.22 26.25 26.25 26.27 26.37	20.32 22.27 22.35 22.41 22.34	22.35 22.38 24.34 24.42 24.42 4.43 4.43
	HOAP	1 999 8 1 1 999 8 1 1 1 9 9 8 8 9 9 8 1 1 1 1	11198 1198 1198 1198 1198	11.131 11.149 11.198 0.951	11.049	11.05	0.949 0.998 1.051 1.132	1.08 1.030 1.038 1.0849	1.0231 1.0231 1.0230 1.0230 1.029
	4E Y 1	0.191 0.173 0.176 0.181	0.193 0.195 0.176 0.183	61198 61198 61198 1176	0.193 0.193 0.195 0.196	6.138 6.188 6.183 2.188	0 173 173 181 198 198 198 198 198 198 198 198 198	0.176 0.176 0.176 0.183	2.193 2.193 2.196 2.193 2.193
	SER	4 4 4 R R	22450	500 000	0 0 0 0 0 0 4 10 0	722	427	989	4 6 8 8 8 8 9 9 4

TABLE 1.5 DATA LISTINGS

	ROLL M	0.0959 0.0868 0.1551 0.1578	0.1143 0.1045 0.0906 0.1651	8.1211 8.1211 8.1182
	YAW A	0.2593 -0.7320 0.2470 -0.7247 0.0748 -0.1803 0.2080 -0.5558	-0.6645 -0.6536 -0.6866 -0.2136	
	SIDE F	0.2593 0.2470 0.2098 0.2098	0.2294 0.2331 0.2362 0.0995	2,3330 0.0376 -0.0245 2,5424 0.2036 -0.5673 2,4780 0.2051 -0.5620
	DRAG	1.8992 1.9189 1.7697 1.8532 2.3286	2.2341 2.1799 2.1785 2.8198 2.1454	2.3338
DE .	BASE	-8.8996 -8.8978 -8.1814 -8.8955	-0.1183 -0.1095 -0.1074 -0.1059	-0.0985 -0.1301 -0.1172
ANGLE = 30 DEL.	AXIAL	-6,6442 -6,6513 -6,2984 -9,3878	-0,6672 -0,6399 -0,6414 -0,2637 -0,3885	-0.5587 -0.6589 -0.6277
ROLL ANG	PITCH	-1.7840 -1.7334 -1.6845 -0.722	-2.0438 -1.9725 -1.9171 -1.6594 -1.0553	-2.2216 -2.1565
	NORMAL	-3.3943 -1.7840 -0,6442 -0.0996 -3.3983 -1.7334 -0,6513 -0.0970 -3.3961 -1.6045 -0,2904 -0.1014 -3.3679 -0,9772 -0,3878 -0.0955 -3.3743 -0.7092 -0,5613 -0.0889	-3.9987 -2.8438 -0,6672 -0.1183 -3.4229 -1.9725 -0.6399 -0.1895 -3.8897 -1.9171 -0,6414 -0.1874 -5.9458 -1.6594 -0,2637 -0.1859 -3.9780 -1.8553 -0,3885 -0.1853	30.2 -4.0420 -0.9684 -0.5507 -0.0985 30.1 -4.3480 -2.2210 -0.6589 -0.1301 30.1 -4.2455 -2.1565 -0.6277 -0.1172 30.1 -4.2455 -2.1565 -0.6277 -0.1172
	RANG	386.2	3388	388
	THETA	24. 44. 44. 64. 64. 64. 64. 64. 64.	26.56 26.51 26.51 28.51 28.55	28.61 28.55 28.56
	HOAP	1.236 6.951 1.631 1.649	11122 11148 11198 1998	1,151
	REYN	0,196 1,149 0,198 1,280 0,178 2,951 0,185 1,031	97 0.191 1,172 98 0.198 1,148 99 0.201 1,198 100 0.181 0.950 101 0.185 0.998	8.198 1.852 8.195 1.181 8.199 1.149
	SER	00000	97 98 99 100 131	102

TABLE 1.6 DATA LISTINGS

	ROLL M	0.0013 0.0013 0.0014 0.0017 0.0017	0.0017 0.0017 0.0012 0.0018	9.9821 9.9817 9.9814 9.9811	0.00010 0.00010 0.00010 0.00013	60.0000 60.0000 60.0000 60.00018 60.00018	0.0002 0.0001 0.0026 0.0017 0.0039	6.0000 9.0000 9.0006 9.0006 9.00046	3.0071 9.0005 9.0018 9.0038	
	YAW A	0.0036 0.0263 0.0265 0.0139	-0.0757 -0.00009 -0.0010 0.0217	-0.2029 -0.0347 -0.0044 0.0021	-0.2003 -0.2000 -0.3062 -0.3062	0.0184 -0.0157 -0.0259 -0.0812 -0.0364	-0.0263 -0.0319 -0.0422 -0.0422 -0.1008	-0.2817 -0.2586 -0.2211 -0.2832 -0.3721	8.115 8.125 8.374 8.325 8.36	2052.2-
	SIDE F	0.0023 0.0026 0.0034 0.0134	0.0273 0.0012 0.0027 -0.0040 0.0055	8.9126 8.9111 8.9829 8.9842	0.2112 0.3135 0.3236 0.3058 0.0058	0.7007 0.0154 0.0151 0.0273	3.0119 0.0047 0.0216 0.0232	0.0294 0.0223 0.0135 0.0341	. 338 . 327 . 321 . 331	0.0287
	DRAG	0.2982 0.3108 0.4731 0.5820	0.5942 0.2767 0.3033 0.4816	0.5964 0.5882 0.2840 0.3151	0.5858 0.5948 0.5924 0.3084	6.5889 8.6885 8.6896 8.6158	0.3475 0.5382 0.6361 0.6435	0,3639 0,3947 0,5776 0,7881 0,6962	7444 61244 1244 1244	6/5/0
DEG.	BASE	0.0000 -0.0122 0.0083 -0.0141 -0.0471	-0.0557 -0.0210 -0.03099 0.0103 -0.0080	-0.0429 -0.0532 -0.0161 -0.0120	-0.0509 -0.0509 -0.0593 -0.0217	-8.8282 -8.8531 -8.8531 -8.8595	-0.0212 0.0020 -0.0363 -0.0587	-0.0211 -0.0230 0.0230 -0.0209 -0.0416	225 225 225 235 235 235 235 235 235 235	-2.0/40
E = 37,5	AXIAL	-0,2935 -0,3179 -0,4600 -0,5899	-0.6439 -0.2978 -0.3132 -0.4714	-0,6393 -0,6415 -0,2942 -0,3205	-0,5955 -0,6385 -0,6443 -0,2984 -0,3152	-0,4738 -0,6189 -0,6347 -8,6583 -0,2845	-0.3125 -0.4765 -0.6116 -2.6408	-0,2841 -0,3139 -0,4814 -0,6363	00.00 00	6660.0-
OLL ANGL	PITCH	0.1288 0.1463 0.2091 0.1106	0.1776 -0.0050 -0.0120 -0.0269	-2.0519 8.0008 -0.1436 -0.1758	-0.1178 -0.2089 -0.1564 -0.3087	-0.1857 -0.3132 -0.4114 -0.3297	-8.4628 -9.2612 -9.5529 -8.5691	-0.7319 -0.6766 -0.3811 -0.8259 -0.7692	0.720 0.979 0.878 1.878	-0006
œ	NORMAL	6 11 4 1 8 6 1 1 4 1 8 8 1 1 4 1 8 9 9 8 1 1 8 9 9 8 8 9 8 8 8 8 8 8 8	0.1814 -0.6150 -0.6188 -0.6239	-2.0361 -2.0076 -0.1755 -3.1938	-8.1823 -0.2191 -0.1931 -8.3479	-0.5459 -0.5875 -0.4192 -0.5912	-8.56996 -8.56996 -8.5146 -8.5146	-2./399 -2./619 -3./119 -3.8587 -3.8587	0.020 0.960 0.903 1.101	-1.0595
	RANG	337.75 5.75 5.75 5.75 5.75	33333 555 555 555 555 555 555 555 555 5	88888 8777 8777	33775	337.75	337. 47.75 47.75	75 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5	1	21.5
	THETA	22222	00000	20000 20000 12000 11000	00044	44440 30000 80000	66666 66666 66666	88888 2000 8000 8000 8000	22000	10.03
	MACH	6.952 1.001 1.050 1.100 1.151	1.199 0.948 1.051 1.130	1.2220 1.2220 1.2220 1.2221	1.128 1.148 1.199 0.950	1.049 1.153 1.198 0.949	1.838 1.849 1.131 1.151	0.950 1.631 1.649 1.132 1.149	40004 .	7011
	REYN	0.178 0.178 0.178 0.183	0.188 0.175 0.175 0.176	0.185 0.178 0.178 0.175	0.135 0.138 0.198 0.178	0.136 0.136 0.198 0.191 0.176	0.180 0.185 0.188 0.191 0.196	0.176 0.181 0.185 0.196	94444	0.1.0
	SER	1110	411 611 711 811	1119 128 121 122 123	124 125 126 127 128	138 138 131 132	136 137 138 139	1444 1444 1444 1444	44444 67 8 8 8 8 8	121

TABLE 1.6 DATA LISTINGS

	ROLL M	8.0090 6.0014 6.0041 6.0033	0.000 0.000 0.000 0.000 0.001 0.001 0.001 0.001	8.0178 8.0128 8.0192 8.0869	0.0244 0.0274 0.0210 0.0223	69.66 69.66 69.66 69.66 69.66 69.66 69.66 69.66 69.66 69.66	6.0883 6.0883 6.08611 6.08615 6.0883	0.0354 0.0922 0.0676 0.0784	6.0451 6.1816 6.1816 8.8918	98.98
	Y AW M	-0.1400 -0.1506 -0.3740 -0.0423	-0.1029 -0.1528 -0.1949 -0.3901 -0.0806	-2.1776 -0.1599 -0.1897 -0.2529	-0.1862 -0.2854 -0.2775 -0.3016	-0.3716 -0.2854 -0.4019 -0.4414	-0.1376 -0.4679 -0.1292 -0.4776	-0.4378 -0.3104 -0.2613 0.1631	-0.4545 -0.4317 -0.0357 -0.1278 0.2011	-0.4186
	SIDE F	0.0575 0.0575 0.0575 0.0195	60000 40000 40000 40000 40000 40000	8.2664 8.2664 8.2652 8.2652 8.2652 8.2652 8.2652	0.0679 0.1094 0.1265 0.1265	0.1110 0.11110 0.11419 0.1892	0.1595 0.1735 0.1735 0.1734	0.1556 0.0182 0.1040 -0.0372	0.1767 0.1538 0.08254 0.0651	0.1530
	DRAG	0.7789 0.4938 0.5321 0.7179	6.858 6.5734 6.6433 6.828 8.843 8.843 8.843	0.9743 0.9536 0.9600 0.6892	0.9284 1.1086 1.0754 1.0977 0.8178	0.9122 1.0787 1.2609 1.2329 1.2597	0.9963 1.1253 1.2674 1.4669	1.1940 1.1940 1.3161 1.4691	1.6333 1.6463 1.4520 1.5627	1.9440
. 046.	BASE	-0.0741 -0.0186 -0.0273 -0.0073	-0.0760 -0.0796 -0.0262 -0.0259	-0.0730 -0.0760 -0.0817 -0.0388 -0.0310	-6.0074 -0.0681 -0.0760 -0.0742	-0.0125 -0.0125 -0.0781 -0.0827	-8.0665 -0.0465 -0.0359 -0.0922 -0.09022	-0.0872 -0.0712 -0.0691 -0.0565	-6.0887 -0.0875 -0.0893 -0.0769	-0.1010
LE = 37,5	AXIAL	-0,6768 -0,2726 -0,3118 -0,4915	-0.6667 -0.6814 -0.2697 -0.3246 -0.5061	-0,6706 -0,6703 -0,6818 -0,2847 -0,3325	-0,5008 -0,6723 -0,6690 -0,6812	-0,3371 -0,5100 -0,6782 -0,6663	-0,2898 -0,3453 -0,5266 -0,6772	-0.6674 -0.2983 -0.3627 -0.5442	-0,6511 -0,6576 -0,2915 -0,3758	-0,6715
SOLL ANGL	PITCH	-0.9166 -1.2836 -0.9983 -0.6139	-1.3991 -1.3815 -1.3821 -1.6821	-1.4017 -1.1787 -1.1159 -1.4545	-0.5858 -1.4896 -1.2458 -1.1789	-0.9103 -0.5389 -1.5394 -1.2440	-1.1013 -0.8555 -0.3666 -1.5148 -1.3153	-1.2918 -1.1376 -0.7288 -0.3418	-1.4519 -1.4847 -1.2781 -4.7728 -9.3956	-1.7624
	VORMAL	-1.0631 -1.1749 -1.2123 -1.1636	-1.5010 -1.2994 -1.5867 -1.4572 -1.5975	-1.5176 -1.2444 -1.2463 -1.6331	-1.6289 -1.8975 -1.6152 -1.8448	-2.0130 -1.9346 -2.2134 -2.1673 -2.2226	-2.4357 -2.5218 -2.6471 -2.5633	-2.5556 -2.6234 -2.7443 -2.6694 -3.6151	-2.9244 -2.9378 -3.6748 -3.1198	-3.4451
	RANG	37.5 37.5 37.5 37.5 37.5	37.5 37.5 37.5 37.5 4.78	37. 37. 37. 37. 37. 37.	37.5 37.5 37.5 37.5 37.5	37.55	37.5 37.5 37.5 37.5 37.5	37.5 37.6 37.6 37.0 37.0	37.00	37.5
	THETA	10.09 12.06 12.09 12.13	12.12 14.67 14.13 14.15	14.12 14.14 14.15 16.11	16.28 16.16 16.18 16.23 18.15	18.25 18.26 18.21 18.21 18.25	28.29 28.35 28.35 28.29	22.38 22.37 22.37 22.43	22.38 22.39 24.35 24.43	24.42
	YACH	1.198 1.094 1.0948 1.198	1.148 6.452 1.638 1.658	1.132 1.149 1.198 1.951	1.050 1.150 1.149 0.950	0.998 1.048 1.131 1.158 1.258	0.952 1.082 1.052 1.130	1.038 1.038 1.038 1.049	11198 1198 1198 1198 1198 1198	1.131
	REYN	0.199 0.168 0.173 0.178	6.185 6.196 6.196 6.172 6.175	0.188 0.183 0.188 0.168	6.176 0.185 0.185 0.183	0.175 0.176 0.185 0.185	0.170 0.173 0.180 0.185	0.198 0.173 0.188 0.188	0.1191 0.1195 0.1186 0.1188	0.188
	SER	155 155 156 157 158	159 168 165 165	163 169 170 171	172 173 175 175	177 178 179 180 181	183 184 185 185	198 198 191 192 193	194 195 196 197 198	199

TABLE 1.6

DATA LISTINGS

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	ROLL M	0.0502 0.0424 0.0986 0.0927	6.00 6.00 6.00 6.00 6.00 6.00 6.00 6.00	0.0821 0.0664 0.0625 0.0501
	Y AW A	-0.4257 -0.4282 -0.0819 -0.1959	0.1399 -0.3722 0.1467 -0.3694 0.1435 -0.3842 0.0822 -0.1225	0.1701 -0.3187 -0.3402
	SIDE F	0.1665 0.1549 0.0535 0.0945	0.1399 0.1467 0.1435 0.0822 0.1225	-0.0213 0.1281 0.1254 0.1350
	DRAG	1.8951 1.8921 1.7598 1.8277 2.0032	2.2243 2.1563 2.1629 1.9911 2.1181	2.3249 2.5345 2.4628 2.4429
	BASE	-0.0982 -0.1069 -0.1069 -0.0887	-0.1078 -0.1042 -0.0981 -0.1043	-0.1021 -0.1225 -0.1116 -0.1050
אחרר אומרי היים	AXIAL	37.5 -3.3532 -1.6239 -0.6455 -0.0982 37.5 -3.3501 -1.5682 -0.6582 -0.0956 37.6 -3.2749 -1.4426 -0.2887 -0.1009 37.6 -3.2214 -0.8198 -0.3751 -0.0887 37.6 -3.244 -0.6160 -0.5580 -0.0909	37,5 -5,8731 -1,8874 -0,6625 -0,1078 37,5 -3,7688 -1,8083 -0,6341 -0,1042 37,5 -3,7685 -1,7731 -0,6344 -0,0981 37,5 -5,8993 -1,4978 -0,2545 -0,1043 37,6 -3,9047 -0,8199 -0,3781 -0,0929	37.e -4.4291 -0.8874 -0.5517 -0.1021 37.5 -4.3266 -2.1128 -0.6531 -0.1225 37.5 -4.2066 -2.0424 -0.6243 -0.1116 37.5 -4.1674 -1.9565 -0.6157 -0.1050
	PITCH	1.6239 -1.5682 -1.4426 -0.8198	1.8874 -1.8003 -1.7731 -1.4978	-0.8874 -2.1128 -2.3424 -1.9565
	NOMMAL	3.5532 3.5501 3.5749 3.5214	3.4731 3.7688 3.8993 3.9847	4.3266 4.2066 4.1674
	RANG	37.00	37.5	37.6 37.5 37.5 37.5
	THETA	26.55 44.56 26.56 26.56	26.59 26.51 26.53 28.47 28.58	28.62 28.56 28.57 28.58
	AACH	1.198 1.198 1.989 1.0000 1.0000	1.100 1.151 1.200 1.949 1.001	1.049 1.099 1.148 1.198
	REYN	0.193 0.196 0.176 0.181 0.186	0.198 0.195 0.196 0.176	0.188 0.198 0.198 0.199
	SER	2022	2222	218 211 212 213

TABLE 1.7 DATA LISTINGS

DATA LISTINGS ROLL ANGLE = 45 DEG.

ROLL M	6.00010 6.0000 6.0000 6.0000 6.0013	0.00000 0.00000 0.00013 0.00013 0.00000 0.00000	2002	0.0019 0.0019 0.0019 0.0019 0.0019 0.0019 0.0019	0.0007 0.0001 0.0010 0.0010 0.0001 0.0001 0.0001 0.0001
YAW	-0.0003 0.0072 0.0255 0.0051	-0.0658 0.0078 0.0078 0.0215 0.0265 -0.0063 0.0032	88288	0.00014 0.00016 0.00016 0.00016 0.00144 0.00016 0.00016 0.00016 0.00016	0.2027 0.2154 0.2154 0.2074 -0.2076 0.213 0.2042
SIDE F	0.0021 0.0013 -0.0035 0.0066	0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0	88888	0.0054 0.0054 0.0051 0.0253 -0.0253 -0.0248 0.0124 0.0124 0.0124	-0.00000000000000000000000000000000000
DRAG	0.3112 0.4965 0.5959	6,5937 6,2884 6,3198 6,5848 6,5848 6,5941 6,3186	ນທູກທຸກ	0.59938 0.59938 0.6238 0.3196 0.3468 0.5427 0.6425 0.65425	8.3593 8.5934 8.5934 8.6908 8.7899 8.4117 8.4883 9.5583
BASE	-0.0137 -0.0162 0.0163 -0.7166 -0.0504	0.0283 0.0281 0.0221 0.0227 0.0297 0.0297 0.0296 0.0296 0.0296 0.0296 0.0296 0.0296 0.0296 0.0296 0.0296 0.0296 0.0296	. 652 . 653 . 617 . 618	7.00998 -0.0357 -0.0535 -0.0526 -0.0526 -0.0529 -0.0529 -0.0529	-0.0253 -0.0195 0.0061 -0.0597 -0.0549 -0.0193 -0.0040 -0.0040
AXIAL	-0,2998 -0,3222 -0,4754 -0,6059	-0,6462 -0,33131 -0,4769 -0,5945 -0,6372 -0,6487 -0,6487 -0,6487	635 635 316	-0, 4813 -0, 66913 -0, 6693 -0, 2844 -0, 2844 -0, 4876 -0, 4876 -0, 6448	-0,2843 -0,3199 -0,4912 -0,6583 -0,6721 -0,2757 -0,3315 -0,6634
PITCH	0.1392 0.1476 0.1008 0.2383	0.1844 -0.0091 -0.0171 -0.0171 -0.0830 -0.0830 -0.1720	. 123 . 227 . 151 . 286	-0.1768 -0.3157 -0.3157 -0.4694 -0.4694 -0.2415 -0.5516 -0.55089	-2.7183 -9.6426 -9.6426 -9.3697 -0.7712 -0.7000 -0.9508 -0.8179 -1.0847 -1.0847
VORMAL	3.1464 0.1561 0.1475 0.1991 2.1399	0.1833 -0.0172 -0.0172 -0.0190 -0.052 -0.0337 -0.0337 -0.0369	4401.00	-2.3458 -0.34247 -0.4247 -0.5036 -0.5271 -0.5176 -0.5176 -0.5176 -0.5158 -0.5158	-2.7369 -2.7591 -2.7591 -2.0568 -2.0568 -2.0530 -2.7539 -2.7539 -1.1138 -1.1138
RANG	4444 4444 3233	4444 4444 3232 232 2323	uuuuu	44444 44000 2322 2222	44444 44444 4 00400 04000 0 22232 22222 2222
THETA	20222	000000000000000000000000000000000000000	22202	44440 00000 800000 00000 800000 00000 800000 00000	8.05 8.05 8.07 8.07 8.07 10.05 10.05 10.07
4ACH	20.00 20.00	10011 11011 19900 10090 04000 40000	44400	11110 1111 21112 4112 4122 4124 6011 6164 6164 6166 6166 6166 6166 616	8963311111111111111111111111111111111111
REY	60.173 60.133 60.133 60.133 60.133	00000 00000 00000 00000 00000 00000 0000	8 119 8 119 1 1 1 1 1 1 1 1 1 1 1 1 1 1	00000000000000000000000000000000000000	0 1 1 3 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2
SER	0 M 4 M 0	78081 HH	12 13 22 22 22 22 22 22 22 22 22 22 22 22 22	22222 2222 22428 22422	888888 84444 4 46068 98488 4

TABLE 1.7

DATA LISTINGS

ROLL ANGLE = 45 DES.

	ROLL M	3.2039 -3.6020 -6.0012 5.6080 5.6080	-0.0009 0.0026 -0.0027 -3.0038	0.0041 -0.0033 0.0039 -0.0034 -0.0034	-0.0031 0.0049 -0.0015 -0.0132	-0.0145 -0.01111 -0.0155 -0.0155	-8.8159 -8.8186 -0.8122 -8.8828 -8.8831	-0.0028 -0.0106 -0.0106 -0.0073	-0.2054 -2.2051 -2.2120 -0.2072 -3.2085	-3.0029
	YAH	-0.0932 0.0130 0.0238 0.0238	34262	-0.0871 -0.0571 -0.0113 -0.0113	0.0504 -7.0112 -6.0162 -0.0629 -6.0103	-0.0081 -0.0295 -0.0126 -0.0464	0.2008 -0.2198 -0.0002 -0.0185	-0.0753 -0.0114 -0.0187 0.0796 -0.0243	-0.0625 -0.0123 -0.0134	-0.0187
	SIDE F	-0.2310 -0.0029 -0.0093 -0.0095	6.0318 -6.0016 -6.0096 -0.0098	6.0062 6.0155 6.0168 7.0056 -6.0081	-8.0165 6.0165 6.0193 6.0199	0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.0	0.0253 0.01158 0.0114 0.0193	0.0297 0.0220 -0.0125 0.0179	0.020 0.020 0.020 0.020 0.020 0.020 0.035	0.0174
	DRAG	0.7848 0.4919 0.5448 0.7370	0.8449 0.5796 0.6594 0.8289	6.9798 6.9466 6.9627 6.0634	0.9455 1.1109 1.0692 1.1033	0.9286 1.1192 1.2675 1.2336 1.2588	0.9916 1.1415 1.2849 1.4722 1.4252	1.4501 1.2358 1.3541 1.5158 1.6962	1.6507 1.6607 1.4505 1.5770 1.7584	1.9415
	BASE	-0.0702 -0.0260 -0.0244 -0.0011	72223	-0.0757 -0.0727 -0.0737 -0.0336	-8.8011 -8.8768 -8.8729 -8.8759	-0.0313 -0.0662 -0.0831 -0.0831	-8.8465 -8.8373 -8.8213 -8.8946 -8.8871	-0.0824 -0.04583 -0.0456 -0.0456	-8.8918 -8.8878 -8.8594 -8.8526 -9.8684	-0.000
GLE = 45	AXIAL	-0,6788 -0,2775 -0,3212 -0,5829	99699	-6.6776 -6.6621 -9.6778 -9.2758 -9.3758	-0.5183 -0.6823 -0.6638 -0.5885	-0,3544 -0,5327 -0,6853 -0,6665	-0.2754 -0.3585 -0.5219 -0.6878	-0,6766 -0,2831 -0,3682 -0,5519	2000	-0,6781
ROLL AND	PITCH	-0.8876 -1.1799 -0.9618 -0.6118 -1.2839	-1.8961 -1.8488 -1.3593 -0.9558	-1.4826 -1.1417 -1.0928 -1.4383 -0.8243	-0.5549 -1.4964 -1.1718 -1.1429	-0.7711 -0.5817 -1.5339 -1.1676 -1.1567	-1.0862 -0.7148 -0.3867 -1.4879	-1,2486 -1,1117 -7,6946 -3,4393 -1,5817	-1.3538 -1.3439 -1.2148 -0.7344 -0.5169	-1.7835
	NOWHAL	-1.8592 -1.1759 -1.2144 -1.1719	-1.2934 -1.2934 -1.3943 -1.3993	-1.6213 -1.5334 -1.5397 -1.6239	-1.6349 -1.8977 -1.7987 -1.8442 -1.9135	-1.9859 -1.9747 -2.2282 -2.1582 -2.2127	-2.2450 -2.4191 -2.5443 -2.5433	-2.5675 -2.7845 -2.7722 -2.7313 -3.0256		-3.4226
	RANG	4444 V444V	4 4 4 4 4 4 0 4 4 4 2 3 3 2 2	4444 V4V44	40444	44444	4444	44444	4444	4.
	THETA	12.67 12.19 12.19 12.69	12.12 14.09 14.14 14.14	144.14	16.21 16.23 16.23 16.21 18.15	18.24 18.27 18.23 18.27	20.23 20.32 20.36 20.31 20.31	22.35 22.38 22.38 22.43	20000 20000 4404	24.42
	N MACH	1 1 1 1 9 9 6 1 1 1 1 1 9 9 9 1 1 1 1 1	44044	3 1 1 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2	83 1.050 85 1.131 86 1.151 91 1.198 70 0.951	76 1.626 95 1.652 85 1.698 96 1.152 91 1.232	6 1,822 3 1,822 5 1,848 5 1,178	5 1.198 5 8.951 6 8.999 3 1.648 8 1.131	44844	3 1,121
	ER REV	45 6.19 44 6.01 44 8.01 47 6.19 47 6.19 81.8	10000	59 8.193 59 8.195 68 8.196 63 8.178 64 8.173	65 6.18 66 6.18 67 6.18 69 6.19	78 6.17 72 6.18 73 6.19 74 6.19	75 6.17 77 6.18 77 6.18 79 6.19	82 6.19 62 6.17 62 6.17 83 6.13	5 6 6 1 1 5 6 6 1 1 5 6 6 1 1 5 6 1 1 7 1 7 1 7 1 7 1 7 1 7 1 7 1 7 1 7	92 0.18
	S		2.2.2.0.0.							

TABLE 1.7 DATA LISTINGS

	ROLL M	0014	.0043	0051	8628	0015	8018	8900	7700	9922	8780	0028	0022	
	æ	91	0- 1	6- 1	0- 6	6- 6	0- 4	8- 6	6- 6	3 -0	- 6	3 -0.	5 -0	
	Y AW M	-0.042	0.0382 -0.0367 -0.0043	-0.306	0.013	-0.013	-0.327	-0.2569	-0.017	-0.016	0.0113	-0.304	-0.018	
	SIDE F	0.0321 -0.0422 -0.0014	0.0382	0.0293 -0.0067 -0.0051	0.0171	0.0185	0.0331	0.0310	0.0424	0.0326 -0.0163 -0.0055	8.0245	0.3194	0.3291	
	DRAG	1.8699	1.7665	1.8458	2.0335	2,2298	2.1570	2,1719	2.0140	2,1356	2.3813 0.0245 0.0112 -0.0078	2.5261	2.4542	
٠٠٠	BASE	-8.8997	9.0849	-0.0684	-0.0832						-8.8827	-0.1159	-0.1084	
ROLL ANGLE = 45 DEG.	AXIAL	0,6495	-0.2824	.0.3800	-0,5649	-0.6720	-0.6414	-0.6445	-0.2532	-0,3862	-0,5612	-0.6533	-0.6260	
ROLL ANG	PITCH	-1.4988	-1.3680	-0.6798	-0.6939	-1.8569	-1.7849	-1.6941	-1.4698	-8.6992	-1.8317	-1.9884	-1.9295	
	NORMAL	44,9 -3.3895 -1.4988 -8,6495 -8.8997 44,9 -3.5414 -1.4942 -8,6558 -8.8954	-3.2729	-3.5120	-3.2863	44,9 -3.8673 -1.8569 -0.6720 -0.1078	-3.7515	-3.7649	-3.9182	-3.9051	44,9 -4.0977 -1.0317 -0,5612 -8.0827	-4.2970	-4.1818	
	RANG	44	44.	4.0	4.4	44.9	44.0	44.7	44.0	44.9	4.4	4.4	4.0	
	THETA	24.44	26.41	26.49	26.53	26.48					28.59	28.56	28.56	
	HOH	1.151	8.952	866.0	1.051	1.130	1,148	1.230	0.950	102 0.175 0.999	103 0.182 1.049	1.098	1.149	
	SER REYN	0.185	0.170	0.175	0.180	0,185	0.188	0.190	0.170	0.175	0.182	0.185	6.188	
	SER	94	95	96	97	86	66	100	101	102	103	104	105	

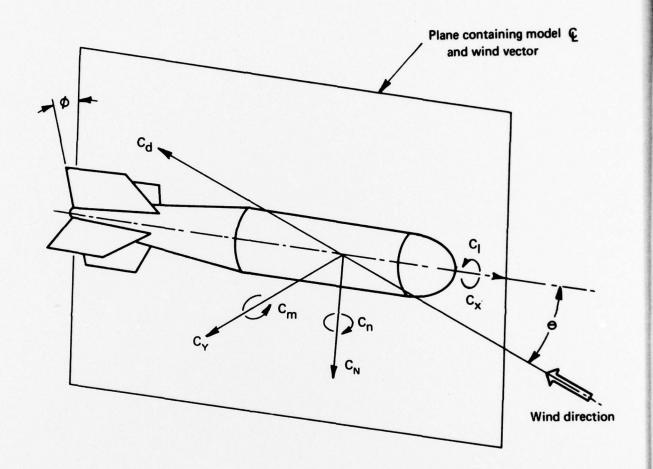
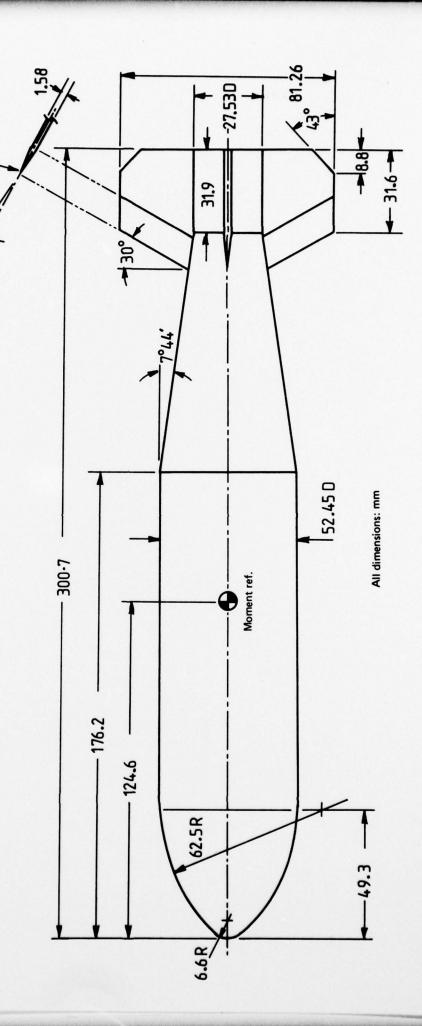


FIG. 1 FORCE AND MOMENT AXES SYSTEM



0.25 radius on leading edge

FIG. 2 DETAILS OF MODEL

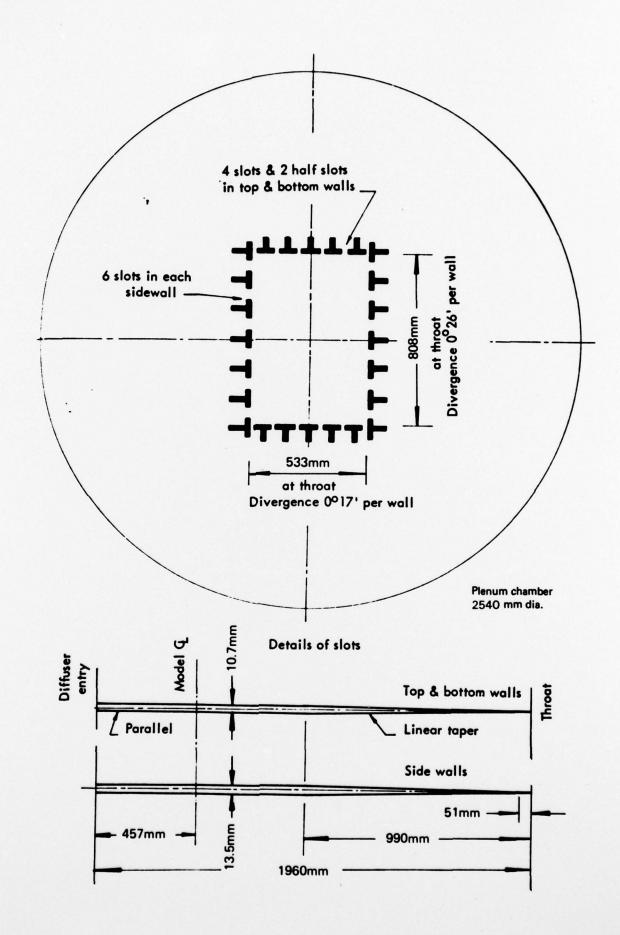


FIG. 3 DETAILS OF SLOTTED TEST SECTION

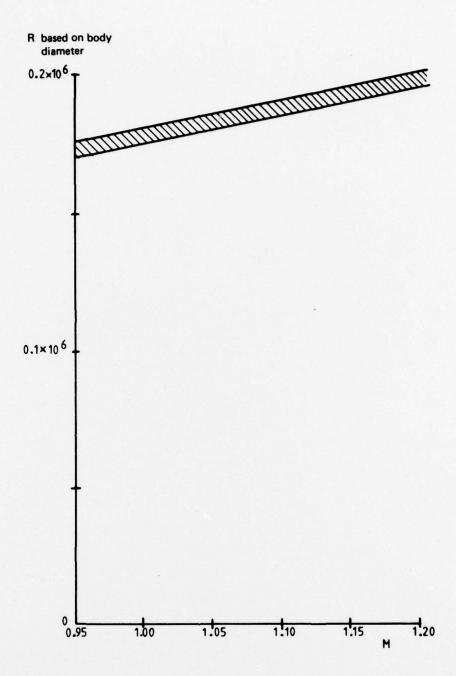


FIG. 4 VARIATION OF TEST REYNOLDS NUMBER WITH MACH NUMBER

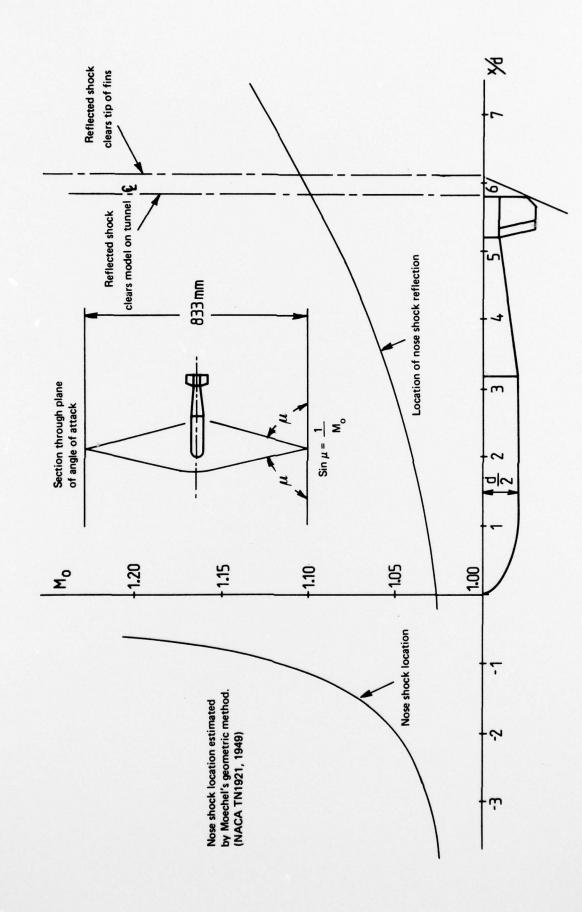


FIG. 5 WALL REFLECTED WAVE INTERFERENCE

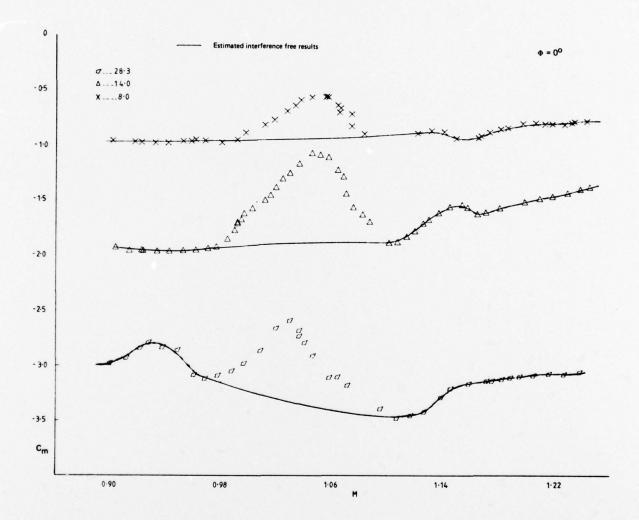


FIG. 6 VARIATION OF $\mathbf{C}_{\mathbf{m}}$ WITH M (FINE M INCREMENT RUNS)

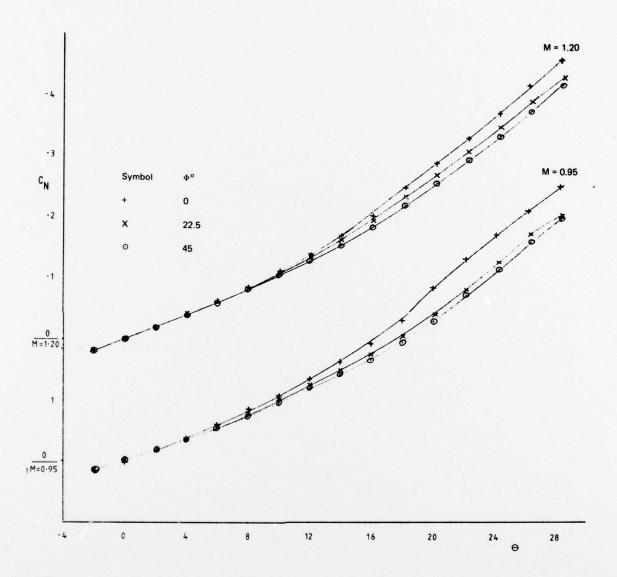


FIG. 7 VARIATION OF C_{N} WITH θ

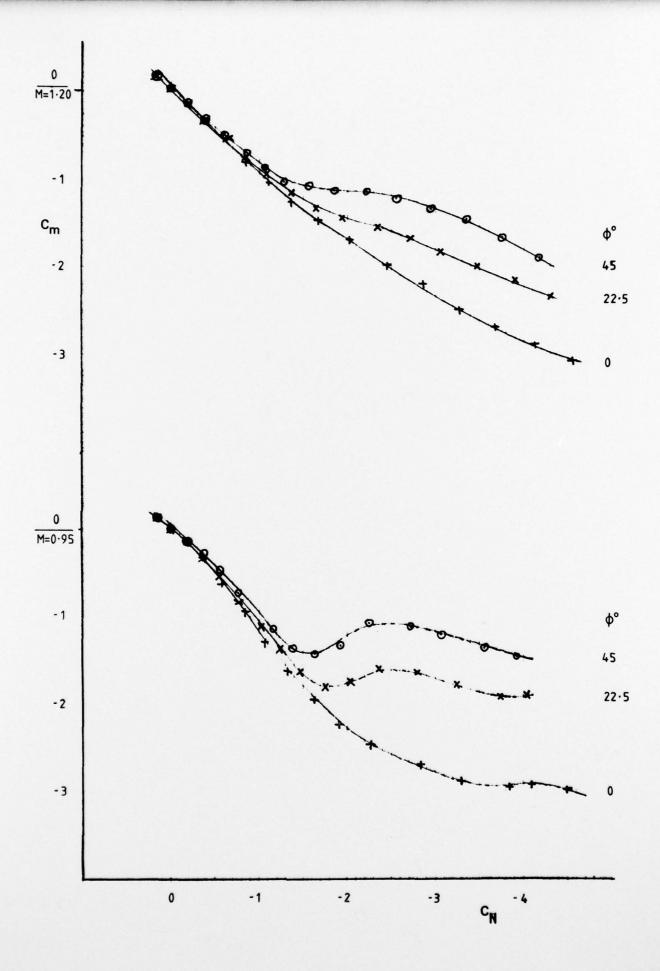


FIG. 8 VARIATION OF $\mathbf{C_m}$ WITH $\mathbf{C_N}$

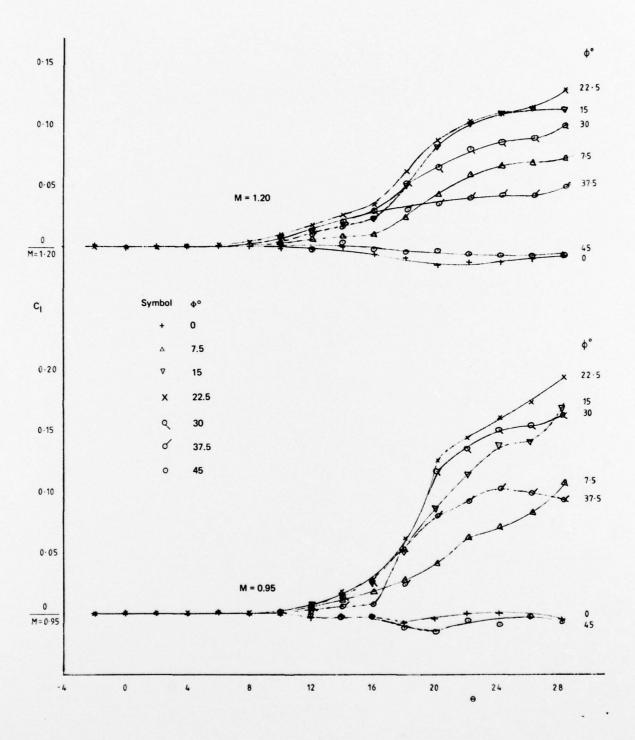


FIG. 9 VARIATION OF C $_{\ell}$ WITH θ

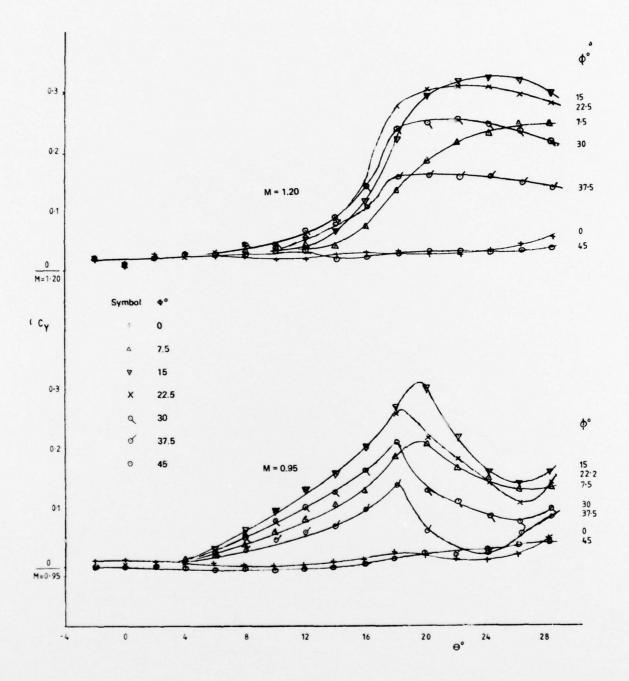


FIG. 10 VARIATION OF $C_{\mathbf{Y}}$ WITH θ

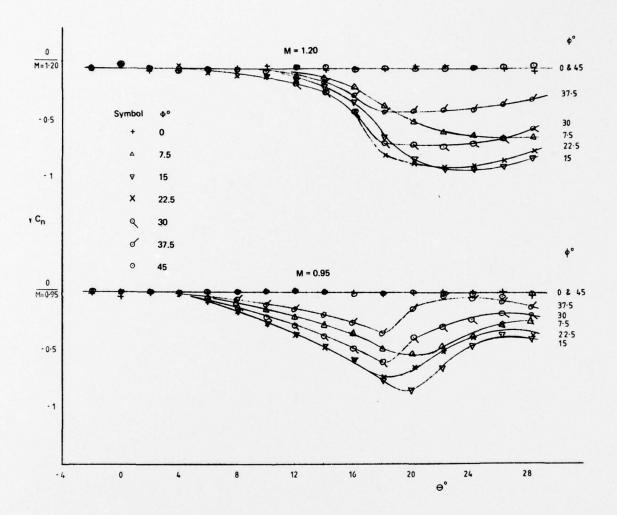


FIG. 11 VARIATION OF C_{Π} WITH θ

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